

IVCHENKO-PROGRESS INNOVATIONS FOR TURBOPROP ENGINES

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October 12th-14th, 2015, Naples

“SCIENTIFIC PRODUCTION ASSOCIATION “A. IVCHENKO” CORPORATION



YEAR OF CORPORATION ESTABLISHMENT - 2007

	1907 - Creation of Motor Sich JSC	1945 - Creation of Ivchenko-Progress SE
Sphere of activity	Development, production and overhaul of gas-turbine engines for civil and military aviation, industrial gas-turbine drives and power-generating plants, consumer goods.	
Structure	14 structural units located in different parts of Ukraine as well as in China, India, UAE, Algeria	one structural unit located in Zaporozhye
Number of employees	over 21 000	3 100



CORPORATION 'SPA 'A.IVCHENKO' ON THE MAP

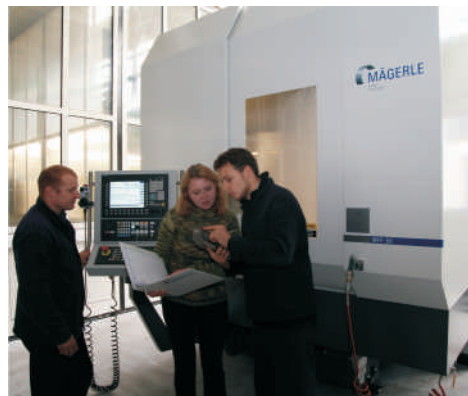


THE BASIC SPHERES OF ACTIVITIES

DESIGN



MANUFACTURE



OVERHAUL



TEST, DEVELOPMENT AND CERTIFICATION



**PUTTING IN SERIES PRODUCTION
AND IMPROVEMENT OF
CONSUMER'S CHARACTERISTICS**



DIRECTIONS OF ACTIVITY

CIVIL AVIATION: commercial helicopters and aircraft



STATE AVIATION: trainers and combat trainers, multipurpose aircraft, military transport aircraft and helicopters



RECOGNITION OF CERTIFICATION AUTHORITIES

Totally 75 certificates of various types



GOSAVIASLUZHBA

*State Department of
Aviation
Transport of Ukraine
Certificates No. SP 004,
No. UA.145.0073, No. 0009 and others*



*Bureau Veritas (France)
Certificate EN 9100:2009 No. FR015515-1,
EN 9110:2009 No. FR015516-1,
ISO 9001:2008 No. UA227484*



*European Aviation
Safety Agency
(Germany)
Certificate No. 216/2008,
No. 1702/2003 Part 21A.23(b)2*

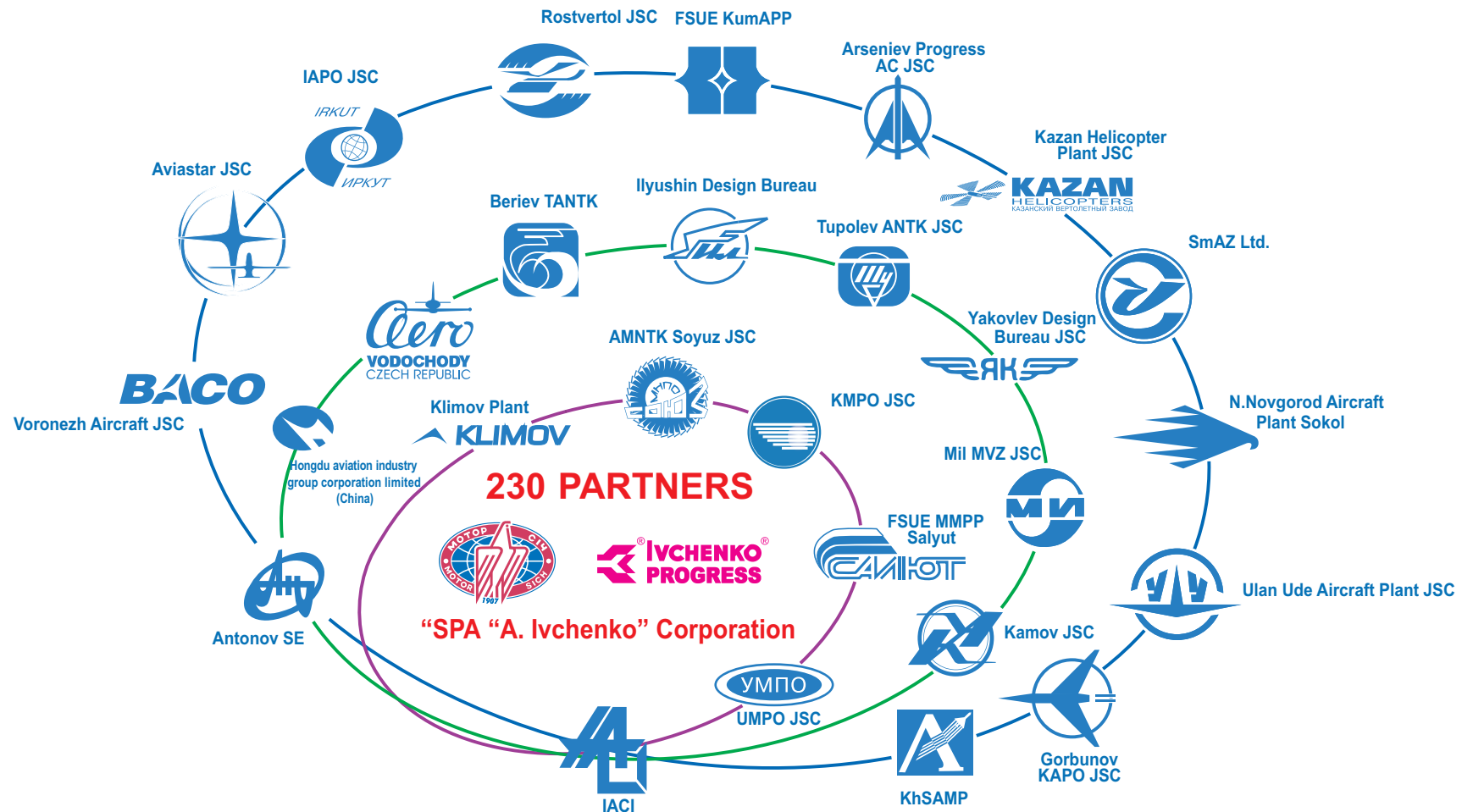


*Aviation Register of
Interstate Aviation Committee
(ARMAK)
Certificates No. SPR-11, No. SPR-15,
No. R-56, No. R-69 and others*



*Federal Air Transport Agency
(Russia)
Certificates No. 2021130360,
No. VR 27.1.4223-2011*

PARTNERS OF “SPA “A. IVCHENKO” CORPORATION DESIGNES AND MANUFACTURERS OF AERO-ENGINES AND AIRCRAFT



Among the partners of “SPA “A. Ivchenko” Corporation are more than **130** designers and suppliers of vendor items and more than **100** suppliers of materials and semi-finished products

COUNTRIES OPERATING AIRCRAFT POWERED BY IVCHENKO CORPORATION ENGINES

IN ALL: 104 COUNTRIES

AZERBAIJAN
ALGERIA
ANGOLA
ARGENTINA
ARMENIA
AFGHANISTAN
BANGLADESH
BELARUS
BULGARIA
BOLIVIA
BRAZIL
BURKINA FASO
BURUNDI
CANADA
CHINA
CHAD
COLUMBIA
CONGO
COTE D'IVOIR
CROATIA
CUBA
CZECH REPUBLIC
CYPRUS
DJIBOUTI
DR CONGO
ECUADOR
EGYPT
EQUATORIAL GUNEA
ERITREA
ESTONIA
ETHIOPIA
GAITI
GEORGIA
GERMANY
GHANA

GREECE
GUINEA-BISSAU
HUNGARY
INDIA
INDONESIA
IRAN
IRAQ
IRELAND
ISRAEL
JAPAN
KAMPUCHEA
KAZAKHSTAN
KENYA
KOREA PDR
KYRGYZSTAN
LAOS
LATVIA
LIBERIA
LIBYA
LITHUANIA
MACEDONIA
MADAGASCAR
MALAYSIA
MALI
MEXICO
MOLDOVA
MONGOLIA
MOZAMBIQUE
NEPAL
NEW ZEALAND
NICARAGUA
NIGER
NIGERIA
PAKISTAN
PALESTINE

PANAMA
PAPUA(NEW GUINEA)
PERU
POLAND
PORTUGAL
ROMANIA
REPUBLIC OF SOUTH AFRICA
RUSSIAN FEDERATION
RWANDA
SENEGAL
SIERRA LEONE
SLOVAKIA
SOMALI
SOUTH KOREA
SPAIN
SRI LANKA
SUDAN
SWITZERLAND
SYRIA
TADJIKISTAN
THAILAND
TURKMENISTAN
TURKEY
WESTERN SAHARA
UGANDA
UKRAINE
UNION OF MYANMAR
UNITED ARAB EMIRATES
USA
UZBEKISTAN
VENEZUELA
VIETNAM
YEMEN
ZIMBABWE

Etc.

Ivchenko-Progress SE Designed Engines

PISTON AND TURBOSHAFT ENGINES FOR LIGHT HELICOPTERS AND ULTRAHEAVY MILITARY TRANSPORT HELICOPTERS



Mi-1



Yak-100



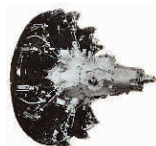
B-10,-11



B-5



G-4



AI-26/GR/GRF
1945*



Ka-26



Ka-18



Ka-15



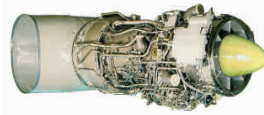
AI-14V
1951*



Mi-26T



Mi-26



D-136
1971*



Ka-226



Mi-2M



AI-450
1994*



MSB-2



Mi-2M/MSB-2



AI-450M,M1
2006*



Mi-26T2



D-136-2
2010*



RUMAS 245



AI-450M2
2013*



Helicopters of take
off weight up to 15 t.



TV3-117VMA-SBM1V
series 5,
TV3-117VMA-SBM2V
2013*



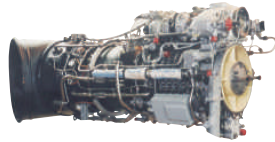
* - year of project launch

ENGINES FOR HELICOPTERS

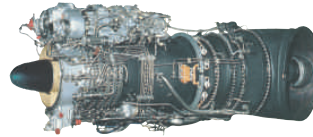
SERIES-PRODUCED AND OVERHAULED ENGINES



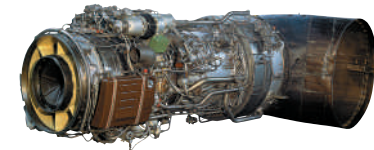
AI-450M
modifications



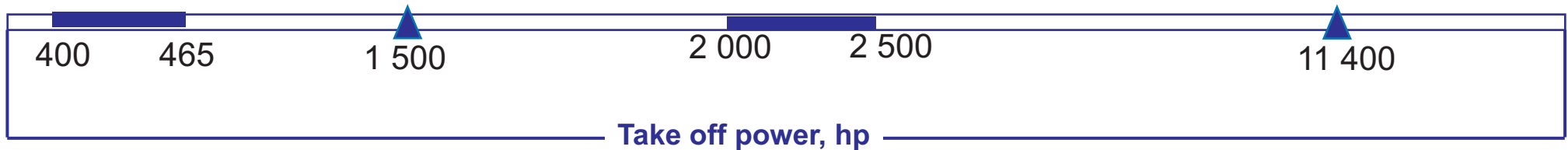
TV3-117VMA-SBM1V
Series 4/ 4E



TV3-117VMA-SBM1V Series 1, 2



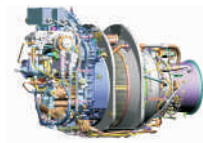
D-136



PROJECTS AND NEW ENGINES



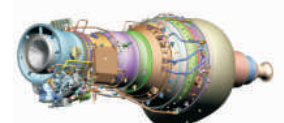
AI-450-2



NEW
TURBOSHAFT



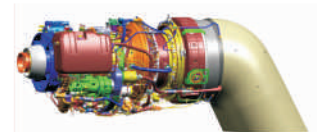
TV3-117VMA-SBM1V
Series 5/-SBM2V



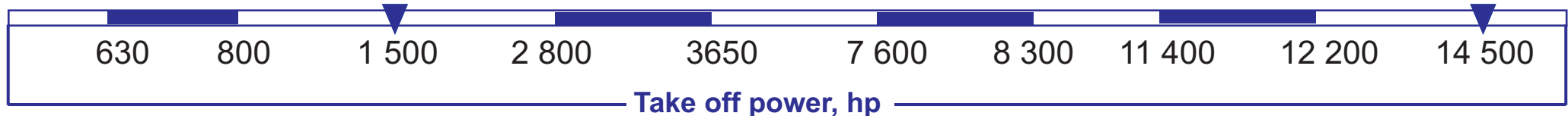
AI-8000V



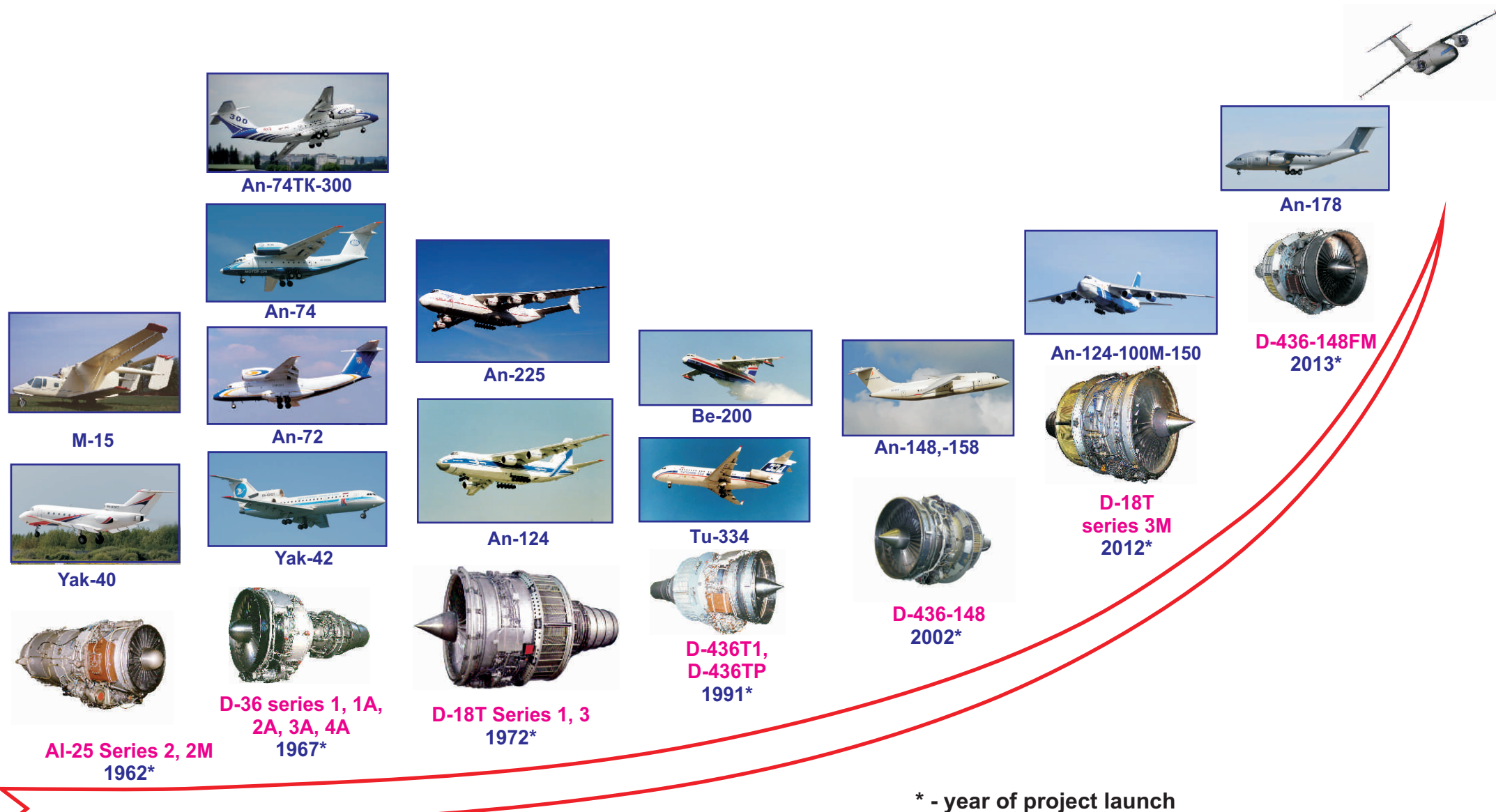
D-136-2/AI-136T



AI-127



TURBOJET BYPASS ENGINES FOR REGIONAL, MILITARY TRANSPORT AND MULTIPURPOSE AIRCRAFT



TURBOFAN ENGINES



AI-450BP/BP-2

AI-25

D-36

D-436

AI-28

D-18T/AI-38

4.0 5.5
(410) (560)

14.7
(1 500)

63.8
(6 500)

62.8
(6 400)

84.4
(8 600)

73.6
(7 500)

98.1
(10 000)

229.9
(23 430)

313.9
(32 000)

Takeoff thrust, kN (kgf)

PISTON, BYPASS TURBOJET ENGINES, INCLUDING ENGINES WITH AFTERBURNER, FOR TRAINERS AND COMBAT TRAINERS



L-39M



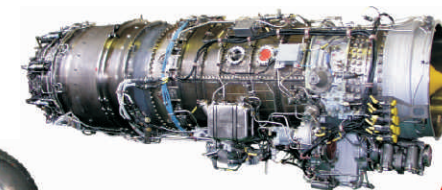
Yak-130D



Yak-130



L-15



AI-222-25F
2005*



Yak-12



JL-8



L-59(39MS)



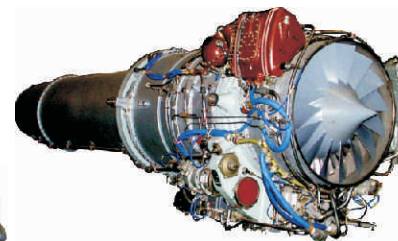
AI-222-25
1998*



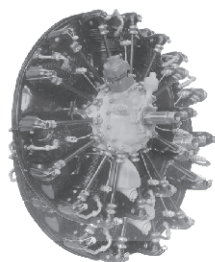
Yak-18



L-39



DV-2
1980*



AI-14
1948*



AI-25TL, AI-25TLK,
AI-25TLSh
1970*

* - year of project launch

ENGINES FOR TRAINERS, COMBAT TRAINERS AND LIGHT COMBAT PLANES



AI-25TL/TLK



AI-25TLSH



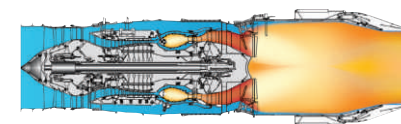
**AI-222-25
modifications**



**AI-222-30
modifications**



**AI-222
versions with
afterburner**



**AI-9500F
engine with
afterburner**

16.9 (1 720)	18.2 (1 850)	24.5 (2 500)	29.4 (3 000)	39.2 (4 000)	49 (5 000)	88.3 (9 000)	98.1 (10 000)
Maximum thrust, kN (kgf)							

TURBOPROP AND TURBOSHAFT ENGINES



AI-450
turboshaft and
turboprop engines



TURBOFAN ENGINES



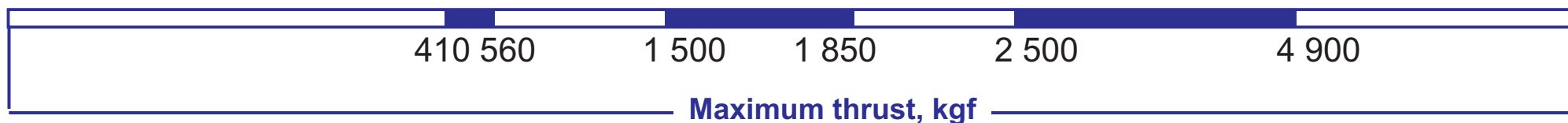
AI-450BP/BP-2



AI-25TL/TLSH



AI-222 modifications



TURBOPROP ENGINES FOR REGIONAL AND MILITARY TRANSPORT AIRCRAFT, SPECIAL PURPOSE AIRCRAFT AND GA



IL-38



IL-20, IL-22



IL-18



An-12



An-10



AI-20A Series 1, 2, 3, 4
AI-20M Series 6
1955*



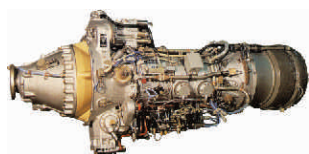
An-30



An-26



An-24



AI-24 Series 2,
AI-24T, AI-24VT
1957*



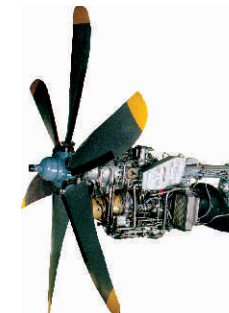
An-70



D-27
1985*



An-140



TV3-117VMA-SBM1
1996*



EV-55



AI-450S2
2011*



DA50-JP7



AI-450S
2012*

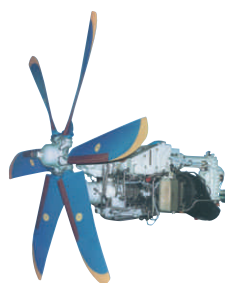


* - year of project launch

TURBOPROP AND TURBOPROPFAN GAS-TURBINE ENGINES



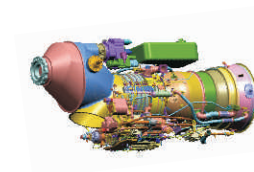
AI-450S/S-2



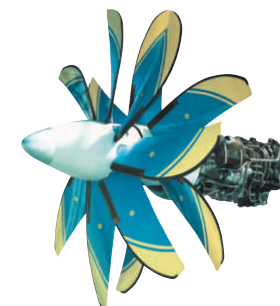
TV3-117VMA-SBM1/2



AI-40



AI-800



D-27

465
(340)

800
(590)

2 500
(1 840)

3 600
(2 650)

3 600
(2 650)

4 000
(2 940)

7 600
(5 588)

8 300
(6 102)

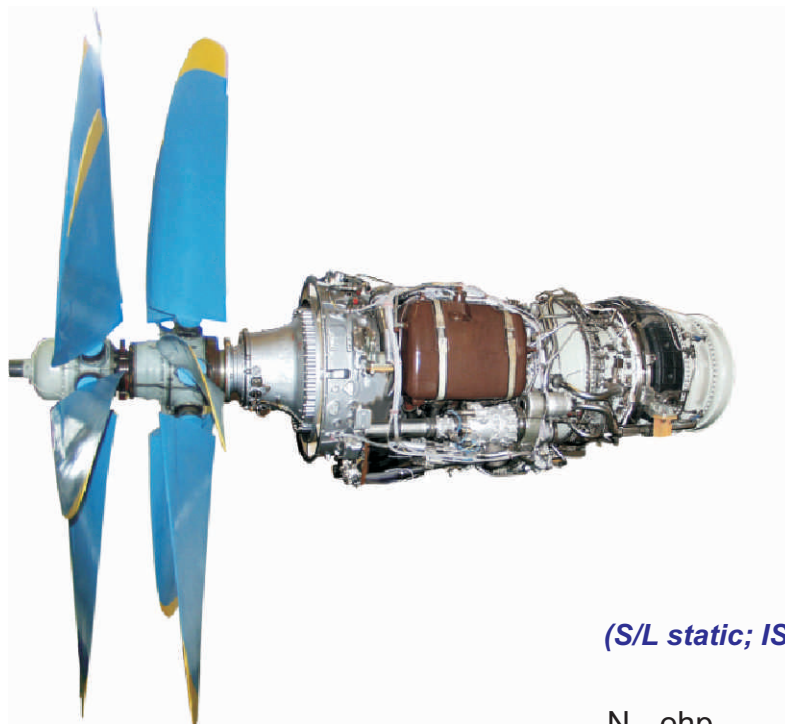
13 240
(9 735)

Takeoff power, hp (kW)

Innovations for Turboprop Engines

D-27 PROPAN

Advanced contra-rotating (open rotor design) engine



(S/L static; ISA)

N_e , ehp

C_N , kg/h/ehp

D-27

Takeoff

13 240

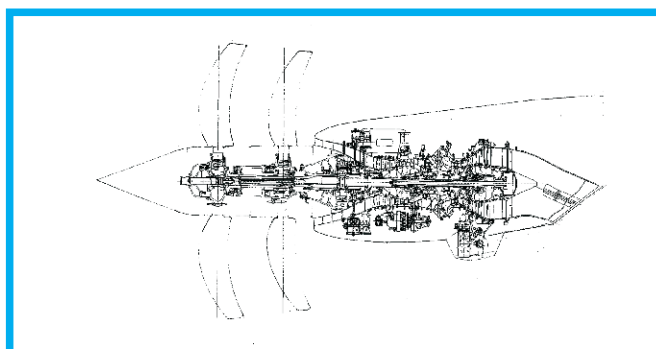
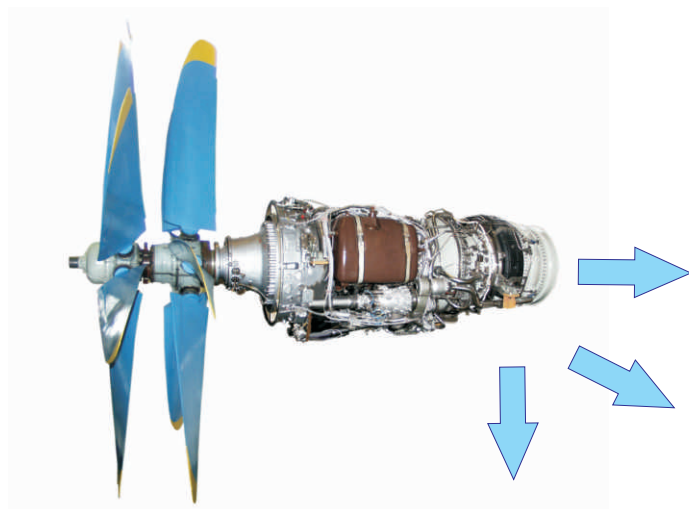
0.180

An-70

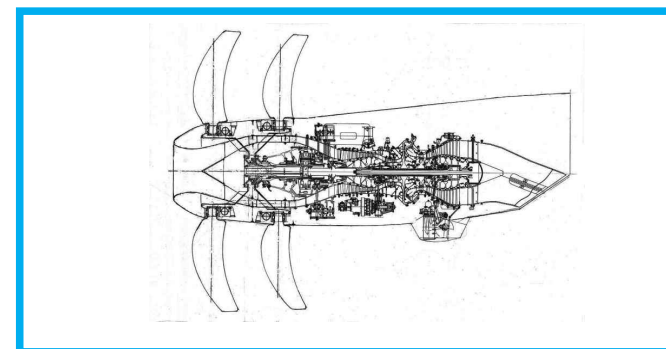


At present activities for mastering the D-27 propfan engine serial production are carried out

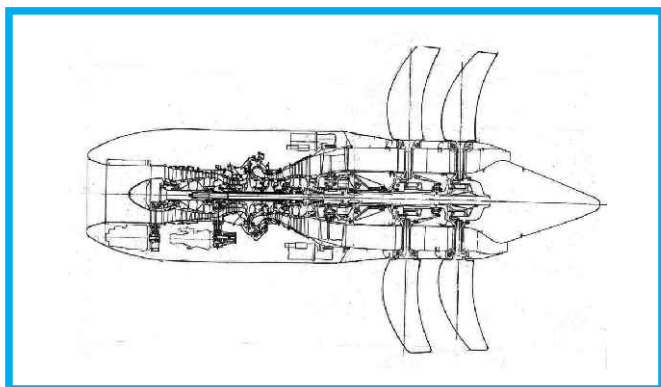
CREATION OF OPEN-ROTOR TYPE ENGINE DEMONSTRATOR



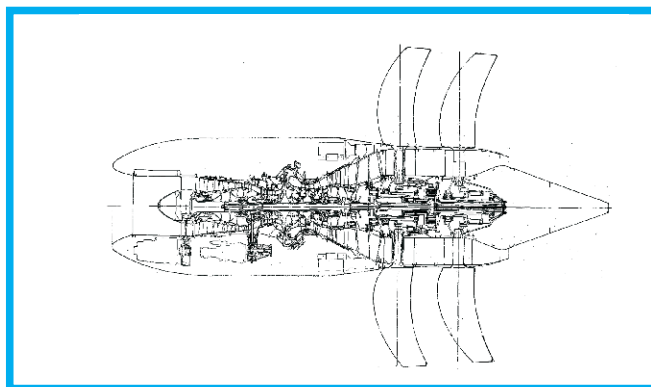
**Tractor propfan of 3.5 m in diameter
with retrofitted air intake**



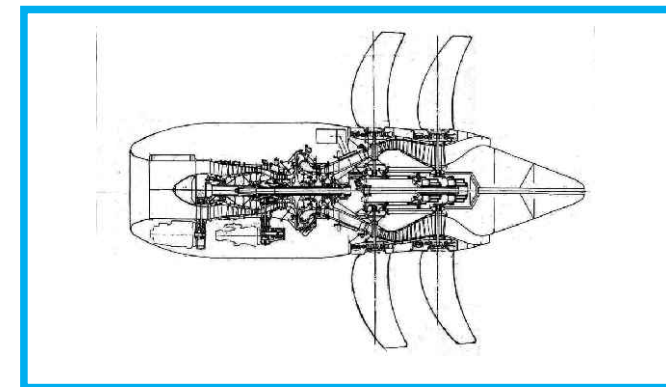
Tractor propfan with forward-type air intake



**Pusher propfan with propfan-rotor
front reduction gear**



**Pusher propfan with reduction gear
between propfan rotors**



Pusher propfan with direct propfan drive

AI-450S, AI-450S-2

Engine can be used on similar aircraft

AI-450S



DA50-JP7 - 19/01/2015 Maiden flight



UAV



Light multi-purpose aircraft

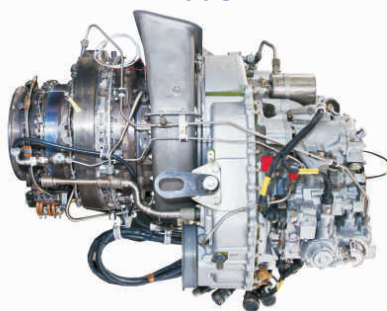


Light trainers



(S/L static; ISA)	AI-450S	AI-450S-2
	Takeoff	
N_{prop} , hp	450...495	630...800
C_N , kg/h/hp	0.277	0.259

AI-450S-2



Developed under **ESPOSA** project
of European 7th Framework Programme

General-aviation light aircraft



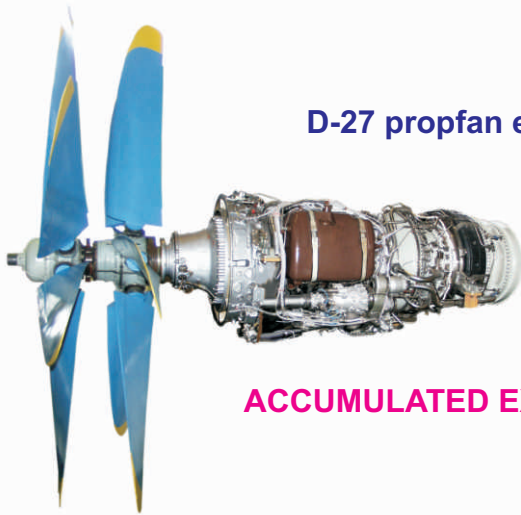
Light multi-purpose aircraft



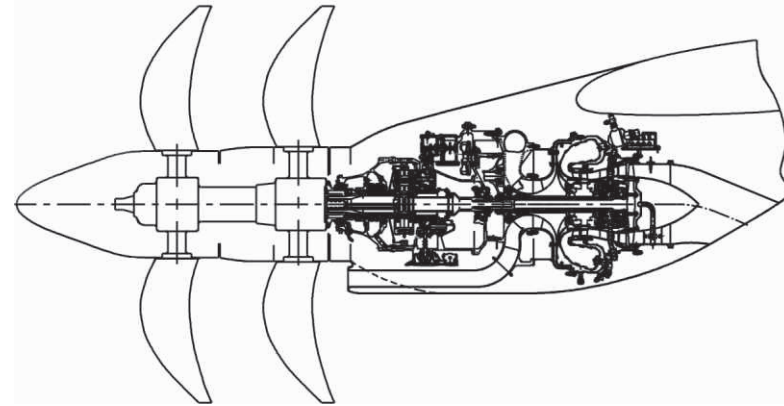
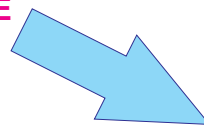
DEVELOPMENT OF ADVANCED GTEs FAMILY CONCEPT (OPEN ROTOR) TO POWER LIGHT AIRPLANES

450...750 hp

D-27 propfan engine

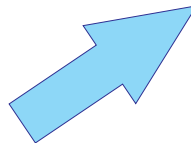


ACCUMULATED EXPERIENCE



Tractor propeller

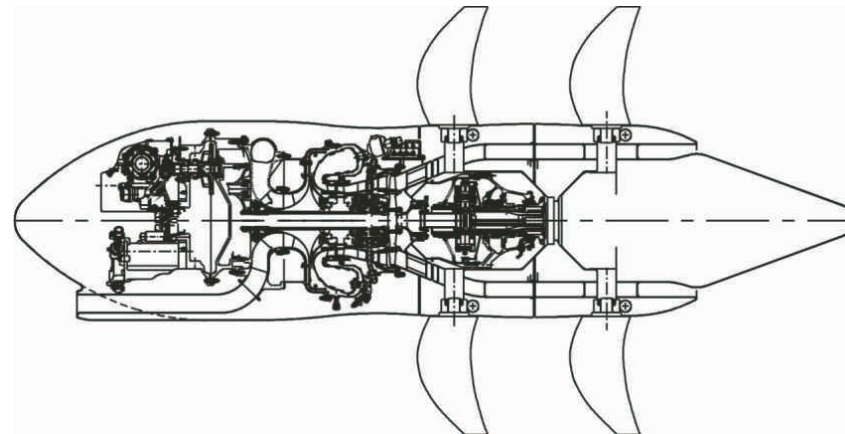
ADVANCED CORE



AI-450S/S-2 turboprop engine



Core developed under **ESPOSA** project
of European 7th Framework Programme

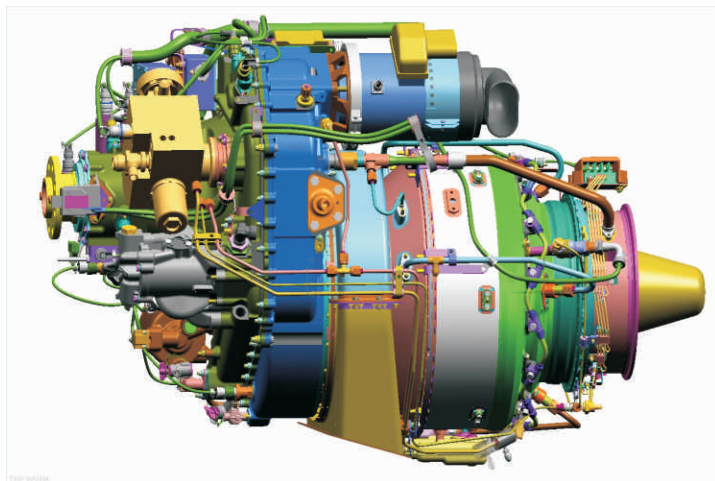


Pusher propeller

ESPOSA - Efficient Systems and Propulsion for Small Aircraft



Starting date - 01/10/2011. Duration in months - 57



Role in the project

SE Ivchenko-Progress will develop the mathematical model and perform trade-off studies for estimation of the performance data for the baseliner engine 2 (BE2) configuration, fulfil the specification and optimization of thermodynamic parameters of BE2 engine in the aircraft-enginepropeller system, develop a design of technical requirements for units and engine systems of BE2 and a design of laws of BE2 engine management, **find an optimal design solution for high pressure compressor, realize the optimization, design and experimental investigations of high efficiency advanced small turbine, fulfil advanced dynamic modelling of high speed turbomachine, supply the specifications of the turbine components coating and carry out the demonstrator validation in terms of coating quality and performance under real service conditions, provider of complex tests and validation of BE2 in test rig, verify mathematical model of BE2 engine with results of tests, investigation of capacity development efficiency of GTE family designs for small aircraft on the basis of single advanced turbocompressor.**



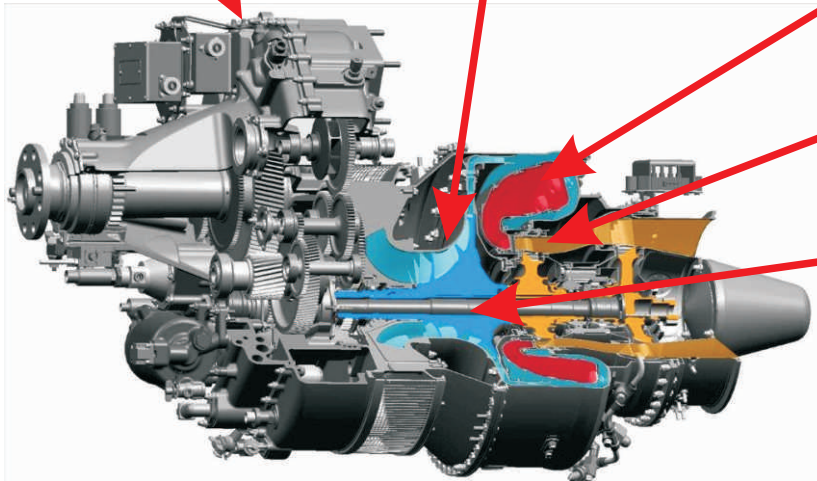
Advanced automatic control system for small engines

Optimal small compressor
(Advanced centrifugal compressor impeller with a new blade geometry- 'winglet')

Efficient combustion concept
(High efficiency, low emission combustor with new designed fuel injector with swirlers)

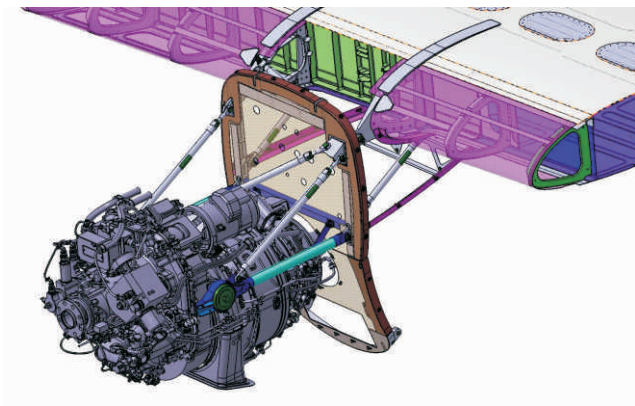
Advanced Cooled Small Turbine
(15 mm size of airfoil)

Advanced dynamic modelling of high speed turbomachinery

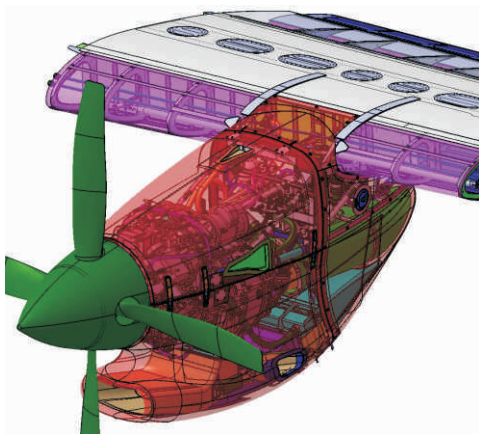




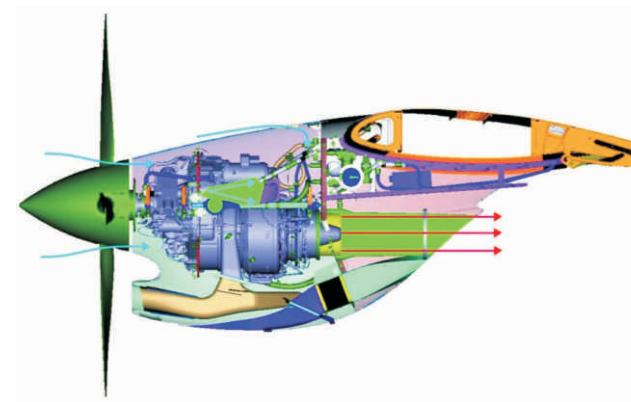
**Complex design methodology for
Engine Mechanical Integration**



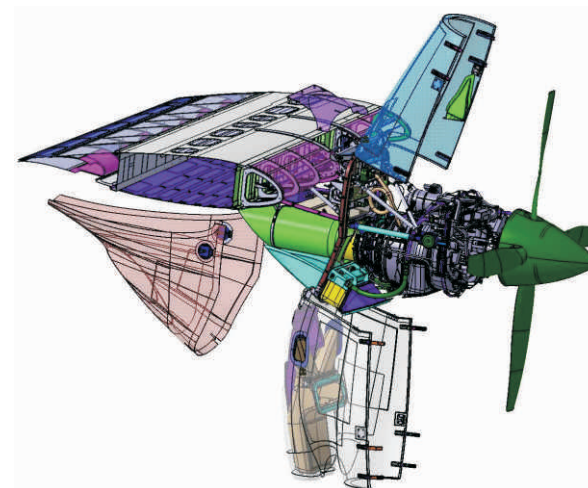
**Reliable design methodology for aerodynamic and
acoustic engine/airframe integration**



**Engine Nacelle Ventilation. Reliable simulation
tools for engine thermal integration**



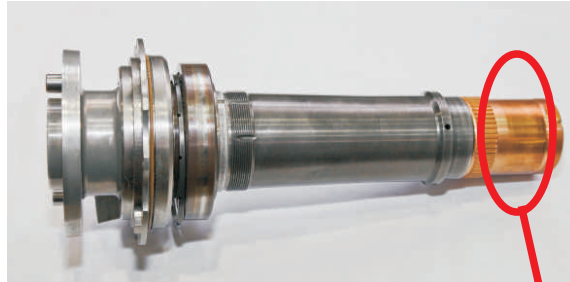
NACELLE. Good accessibility and maintainability



Technology Innovations for Turboprop Engines

DEVELOPMENT OF PROGRESSIVE COATING SOLUTIONS FOR ENGINE PARTS

Wear resistant coatings for bearings
mounting seats on engine shafts



ESPOSA

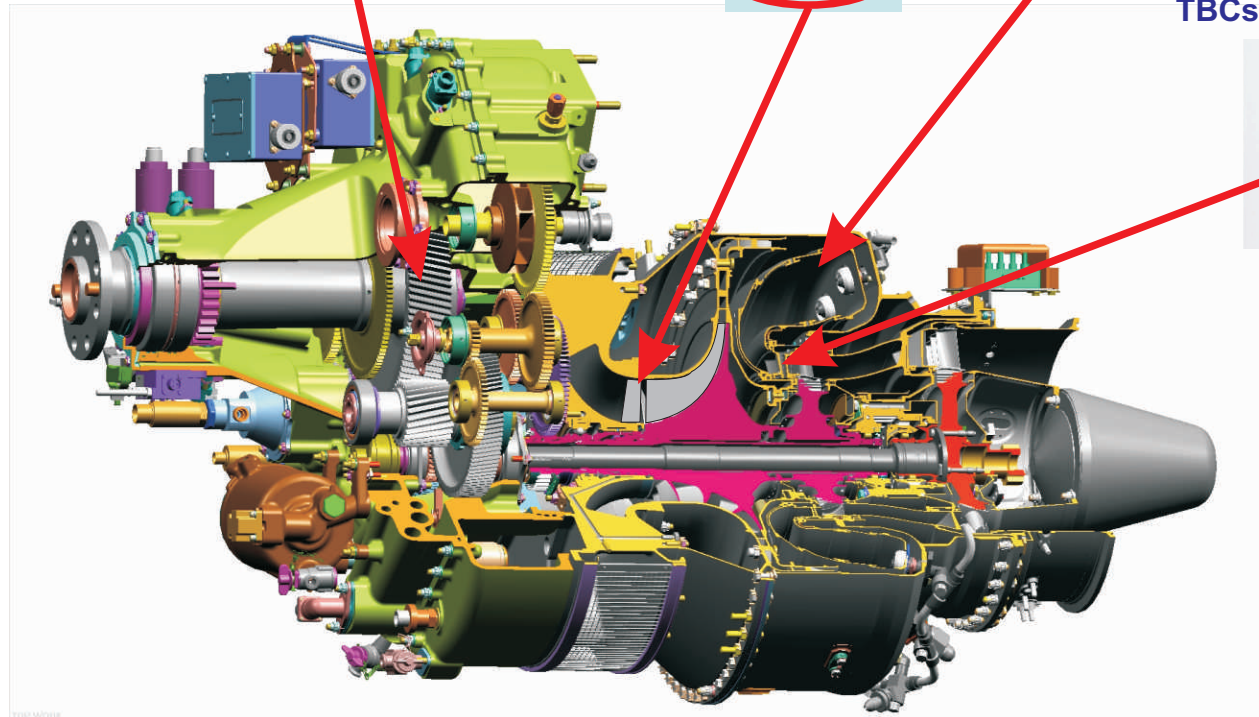
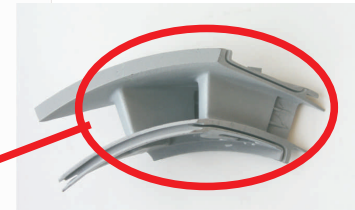
Anti-fretting and anti-wear coatings
for compressor rotor blade roots



Heat-protective coating for internal
surface of small combustor



TBCs for turbine nozzle guide vanes



OXIGEN - Oxide Dispersion Strengthened Materials for the Additive Manufacture of High Temperature Components in Power Generation



Starting date - 01/02/2013. Duration in months - 48

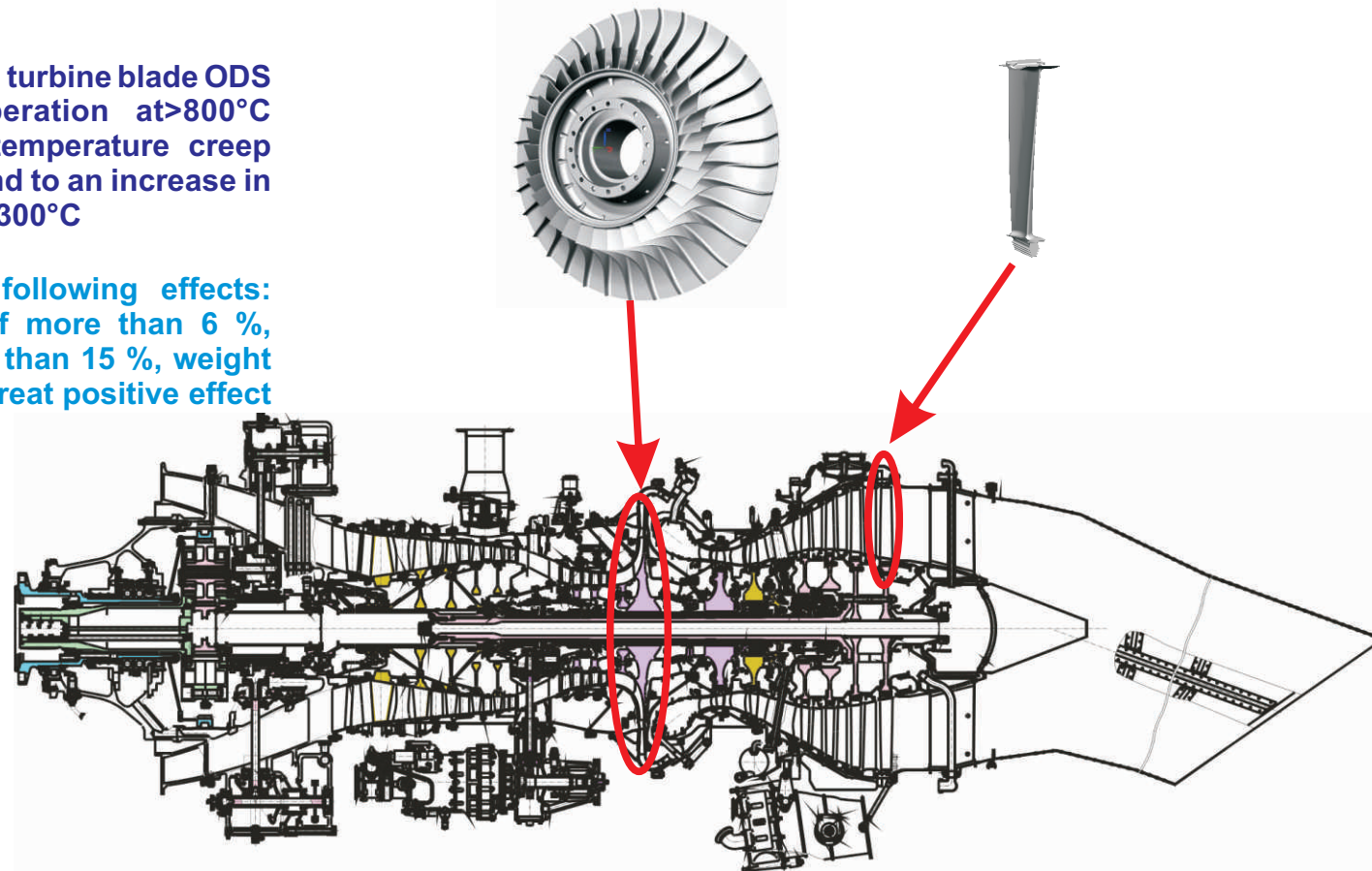
The manufacture of a compressor wheel and turbine blade ODS titanium-aluminium alloy will allow operation at $>800^{\circ}\text{C}$ (theoretically, based on increased high temperature creep strength properties) which would correspond to an increase in operating (inlet/outlet) temperatures by $200\text{-}300^{\circ}\text{C}$

Application of new materials will give following effects: reduction of specific fuel consumption of more than 6 %, increase of power can be reached of more than 15 %, weight reduction, engine life increasing, giving a great positive effect on the engine in whole

ODS TiAl alloy

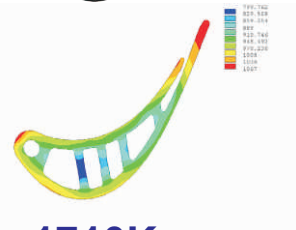
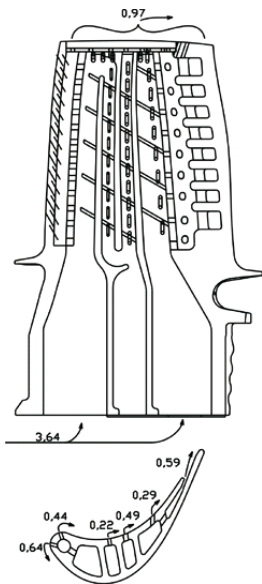
Centrifugal compressor wheel

Turbine blade



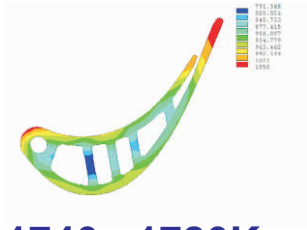
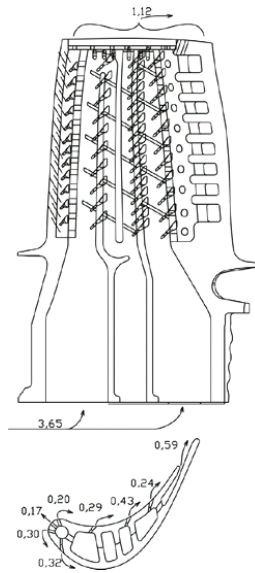
ADVANCED AERO-ENGINE HIGH PRESSURE TURBINE BLADE COOLING SYSTEM CONCEPT

Base design



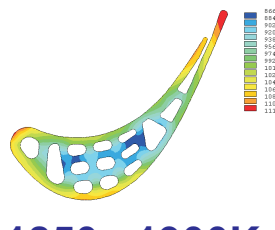
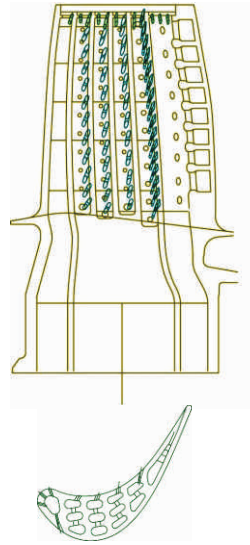
1710K

**Modification 1,
modified blade with
shaped film cooling holes**



1740...1780K

**Modification 2,
wall-cooled system
("penetrating cooling")**



1850...1900K

Blade Mean Section temperatures	Base	Modification 1	Modification 2
Leading Edge, °C	1067	1050	1031
Trailing Edge, °C	1063	1045	1015
Bulk Metal, °C	943	920	875

Modification 2, wall-cooled system blades has an average temperature of the middle section 68°C lower than the blade with base design.

Additionally, the use of TBC with 0.14 mm thickness and a thermal conductivity 2.8 W/(m•K) leads to the decrease in the average temperature of the blade cross section for 15 ... 20°C for all presented cooling blades.

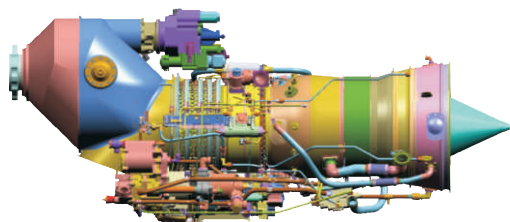
The results of presented investigation show, that one of the most challenging cooling system for the engines with gas temperature at HPT rotor blade inlet TET = 1800...1900 K is a wallcooled system ("penetrating cooling")



New Designed Engines

TV3-117VMA-SBM2, AI-40, AI-8000

TV3-117VMA-SBM2



TV3-117VMA-SBM2

(S/L static; ISA)

	Takeoff	Emergency
N_e , hp	2 800	3 600
flat-rated to $t^\circ\text{C}$	+35	+25
C_e , kg/h/ehp	0.210	

MA60/ MA700 type
Regional airplanes



AI-40



AI-40

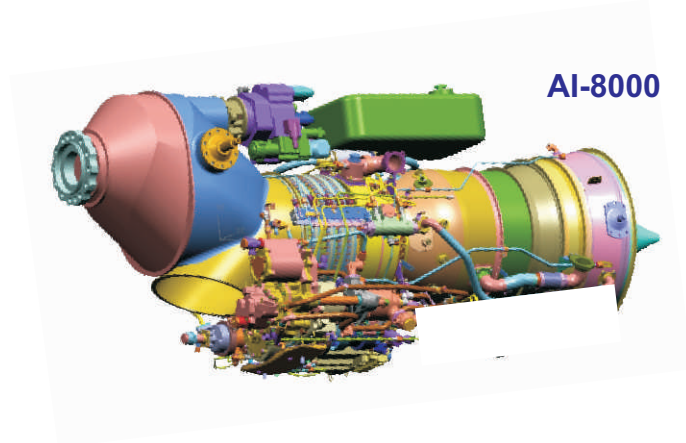
(S/L static; ISA)

	Takeoff	Emergency
N_e , ehp	3 600	4 000
flat-rated to $t^\circ\text{C}$	+40	+40
C_e , kg/h/ehp	0.196	

Regional passenger
and light transport aircraft



AI-8000



AI-8000

(S/L static; ISA)

	Takeoff	Emergency
N_e , ehp	7 600	8 300
C_e , kg/h/ehp	0.175	

Transport and passenger aircraft



OUR MAIN GOALS FOR CUSTOMER SUCCESS

Safety

Reliability

Innovation



Economy

Efficiency

Modernization

We provide power to your fly



ENGINES DESIGNED BY IVCHENKO CORPORATION HAVE BEEN OPERATED IN MORE THAN 100 COUNTRIES OF THE WORLD



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**THANK YOU
FOR ATTENTION!**