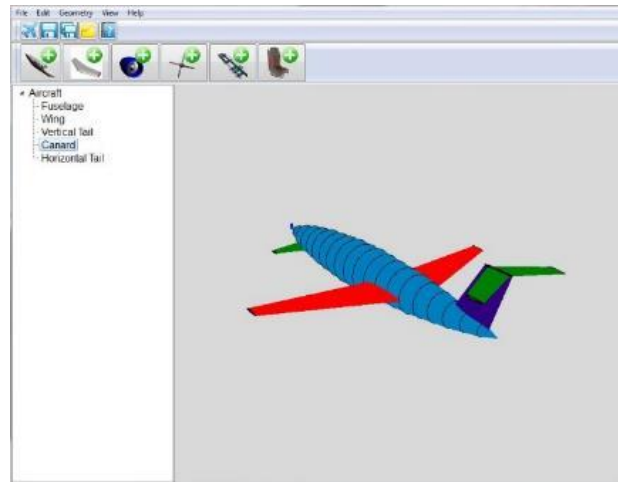


CEAS Technical Committee Aircraft Design (TCAD)
Research Section (SCAD)
October 12-14, 2015
Naples, Italy

Smart Aircraft Modeler (SAM) for Aircraft Conceptual Design



Dr. Willem A.J. Anemaat
anemaat@darcorp.com

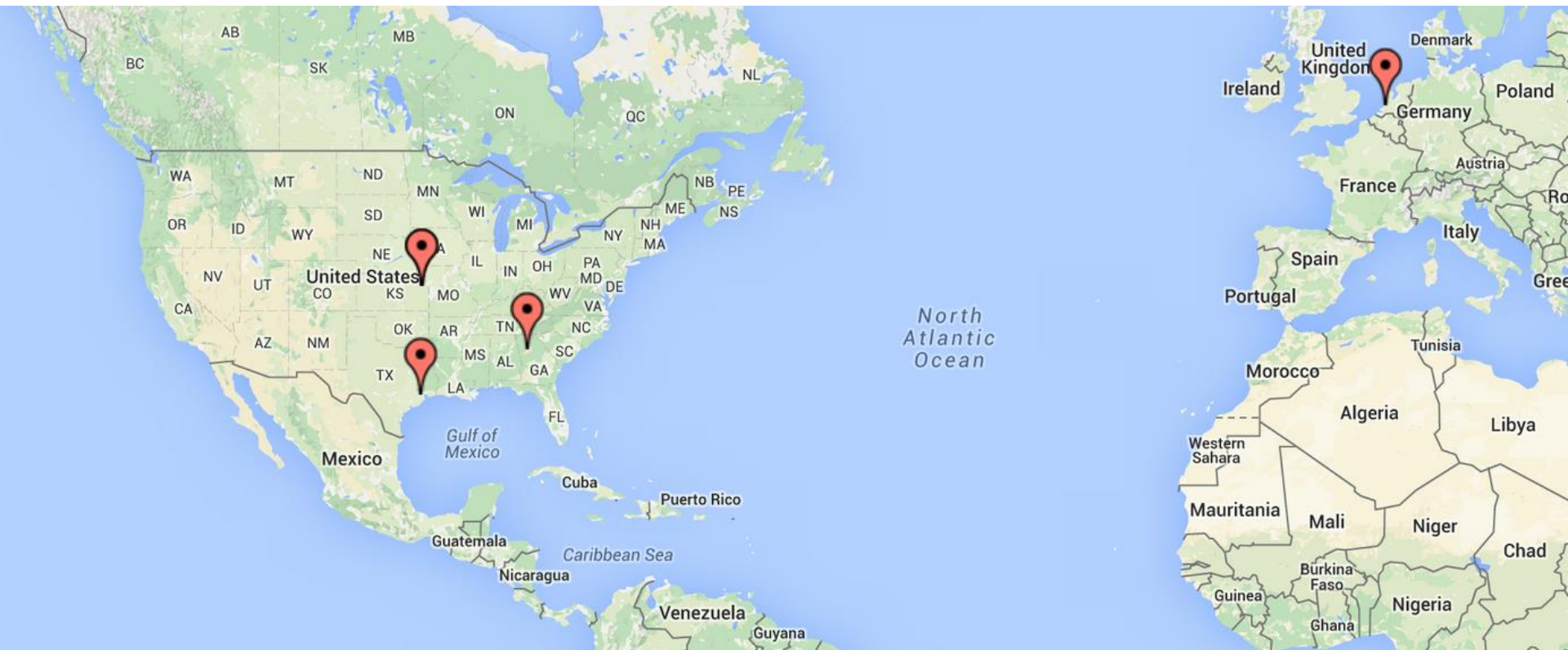
Design, Analysis and Research Corporation

- Formed in 1991 by Dr. Jan Roskam and Dr. Willem Anemaat
- Located in Lawrence, Kansas, USA
- Mission:

Provide Integrated Aircraft Design, Development and Consulting Services

- Market, Support and Develop Airplane Design and Analysis Software
- Airplane Analysis and Design Engineering Consulting Services
- Market and Distribute Airplane Design Books Written by Dr. Jan Roskam
- Wind Turbine Design (Aerodynamics & Structures)

- Headquartered in Lawrence, Kansas
- Branches in Atlanta, Georgia and Houston, Texas
- Coming Soon: The Netherlands





SAM: Smart Aircraft Modeler

- 3-D Parametric Modeling Environment
- Inboard Profile
- Subsystems Layout
- ACL: Aircraft and Component Library
 - 3-D Models
 - Database of Specifications

Advanced Aircraft Analysis 3.6.2 - C525.aaa - Sizing

File Edit Window Airfoil Help

Weight Aerodynamics Performance Geometry Propulsion Stab. & Control Dynamics Loads Structures \$\$\$ Cost

Fuselage Geometry: Sizing

Calculate Plot Clear Out Import Table Export Theory Close

Input Parameters

X_{apex_f}	2.54 m	l_f	0.00 deg	X_{apex_h}	13.91 m	c_{r_v}	2.79 m	N_f stations	27
Y_{apex_f}	0.00 m	X_{apex_w}	7.76 m	c_{r_h}	1.37 m	$(X,Z)_{apex_f}$ Apex is not included			
Z_{apex_f}	2.54 m	c_{r_w}	2.43 m	X_{apex_v}	11.17 m	$(X,Y,Z)_{fus}$ Fuselage Coordinate System			

Fuselage Table: double click for Cross-Section Dialog

Section	Input	Input	Input	Input	Input	Input	Input	Input	Input	Input	Input	Input	Input	Input	Output	Output
1	0.0000	0.0000	0.0002	0.0067	0.0000	0.0000	-0.0054	0.0067	0.0002	0.7080	0.0067	-0.0054	0.7070	0.00	0.0324	
2	0.4467	0.0000	0.3240	0.4796	0.0416	0.0000	-0.2701	0.4796	0.3240	0.8040	0.4796	-0.2701	0.7070	0.48	2.5605	
3	0.8935	0.0000	0.4501	0.6396	0.0929	0.0000	-0.3081	0.6396	0.4501	0.8310	0.6396	-0.3081	0.7070	0.82	3.3998	
4	1.3402	0.0000	0.5532	0.7350	0.1613	0.0000	-0.3363	0.7350	0.5532	0.8350	0.7350	-0.3363	0.7070	1.10	3.9340	
5	1.7869	0.0000	0.6458	0.7885	0.2409	0.0000	-0.3543	0.7885	0.6458	0.8400	0.7885	-0.3543	0.7070	1.32	4.2910	
6	2.2337	0.0000	0.8804	0.8123	0.3131	0.0000	-0.3642	0.8123	0.8804	0.7470	0.8123	-0.3642	0.7070	1.63	4.5928	
7	2.6804	0.0000	1.1652	0.8193	0.3628	0.0000	-0.3695	0.8193	1.1652	0.7200	0.8193	-0.3695	0.7070	1.99	5.0113	
8	3.1271	0.0000	1.2298	0.8219	0.3881	0.0000	-0.3716	0.8219	1.2298	0.7070	0.8219	-0.3716	0.7070	2.07	5.0982	
9	3.5739	0.0000	1.2307	0.8221	0.4023	0.0000	-0.3718	0.8221	1.2307	0.7070	0.8221	-0.3718	0.7070	2.07	5.1003	

Output Parameters

l_f	11.62 m	S_{B_s}	14.48 m ²	S_o	0.89 m ²	Z_{tc_w}	2.97 m	Z_{tc_v}	3.06 m
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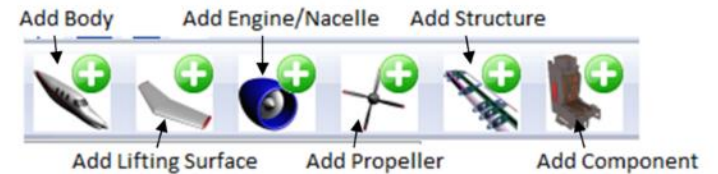
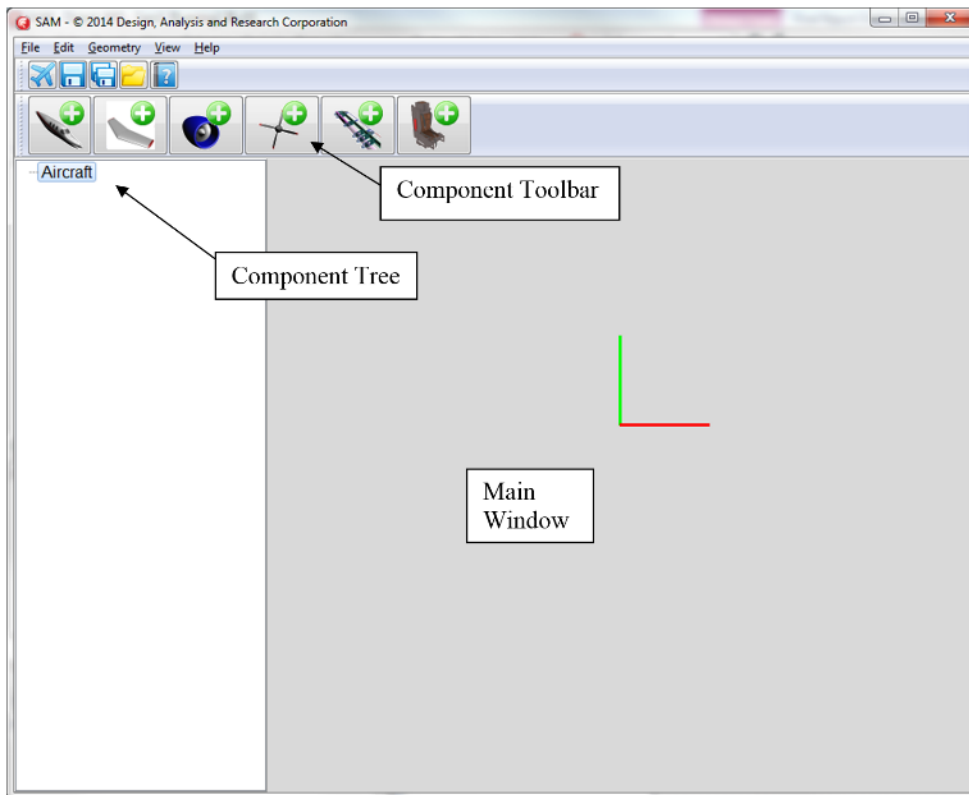
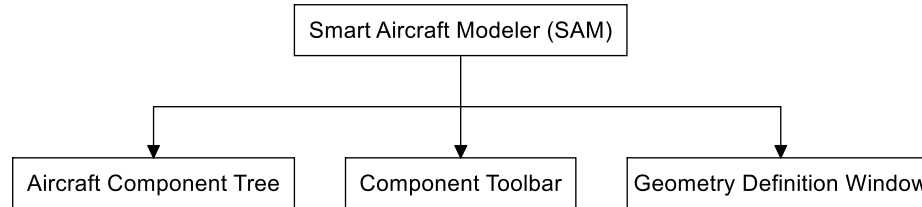
New Open Save Save As Delete

Flight Cond Recalculate Notes Copy WMF Print Atmosphere Help Exit

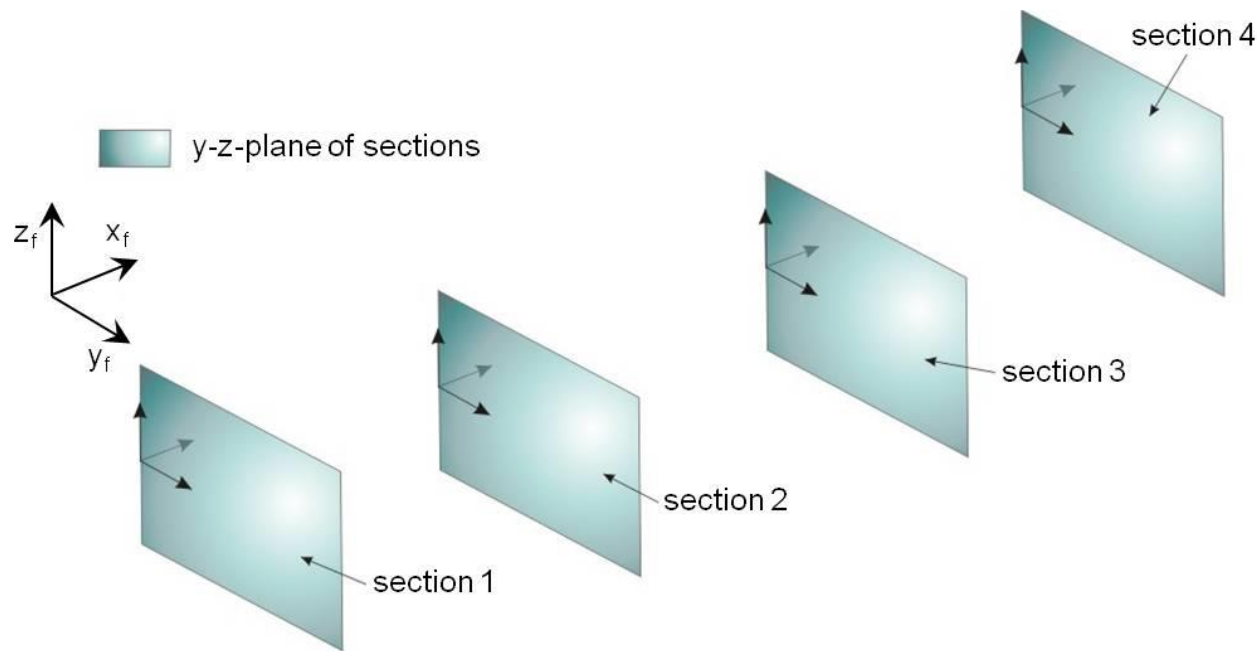
File/Configuration/Certification/Setup

DARcorporation Cessna 525 Citationjet 10/13/15 8:19 AM

SAM Layout and GUI



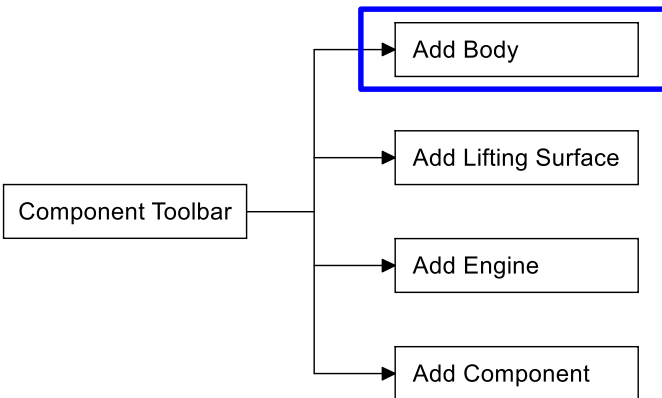
Format: Common Aircraft Configuration Schema (CPACS)



▲	e	sections
▲	e	section
	a	uID
	e	name
	e	description
▷	e	transformation
▲	e	elements
▲	e	element
	a	uID
	e	name
	e	description
▷	e	airfoilUID
▷	e	transformation
▷	e	section
▷	e	section
▷	e	section

Component Toolbar: Body Example

Body Properties Definition



Body1 - General Tab

Name: Body1

Attached to: BodyParentComboBox

Show/Hide: ☒ Show ☐ Hide

Visualization: ☐ Wireframe ☒ Surface

Wireframe Properties: [Color swatch]

Surface Properties: [Color swatch]

Ambient Light Properties: [Color swatch]

Diffuse Light Properties: [Color swatch]

Specular Light Properties: [Color swatch]

Emitted Light Properties: [Color swatch]

Transparency: Transpa [Slider]

Shininess: Edit1 [Slider]

Body1 - Position Tab

Position:

X_{Apex}: 0 [Slider]

Y_{Apex}: 0 [Slider]

Z_{Apex}: 0 [Slider]

i_{Body}: 0 [Slider]

ψ_{Body}: 0 [Slider]

Γ_{Body}: 0 [Slider]

N_{CrossSectPts}: 40 [Slider]

Scale: 1 [Slider]

Mirror: ☒ None ☐ X-Y ☐ X-Z ☐ Y-Z

Body1 - Cross Section Tab

N_{CrossSect}: 4

Cross Section Parameters

Cross Section: 2 [Dropdown]

Cross Section Type: General [Dropdown]

H_{cs}: 5 [Slider]

W_{cs}: 5 [Slider]

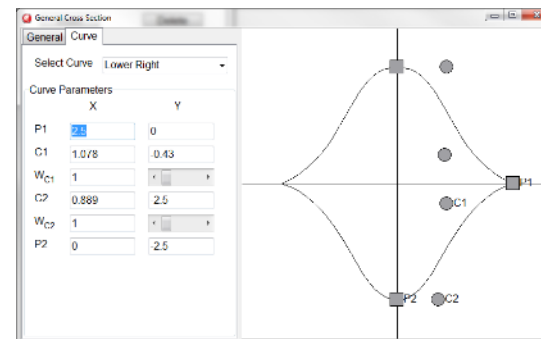
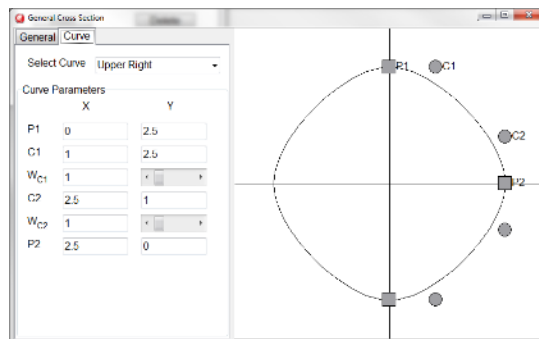
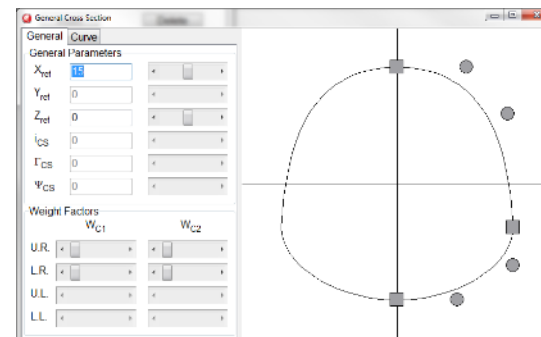
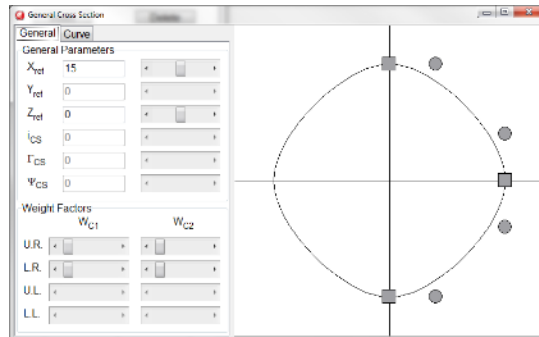
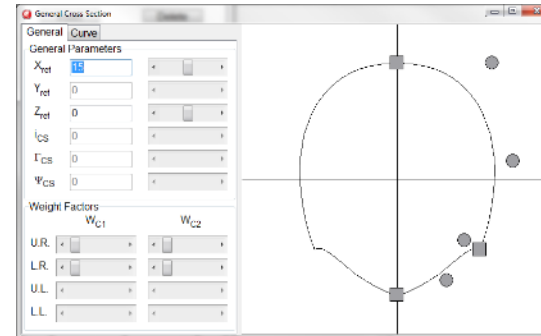
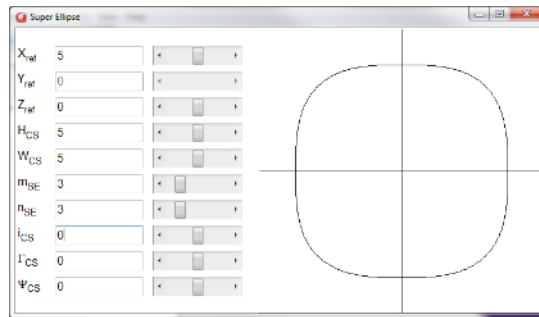
X_{cs}: 5 [Slider]

Scale: 1 [Slider]

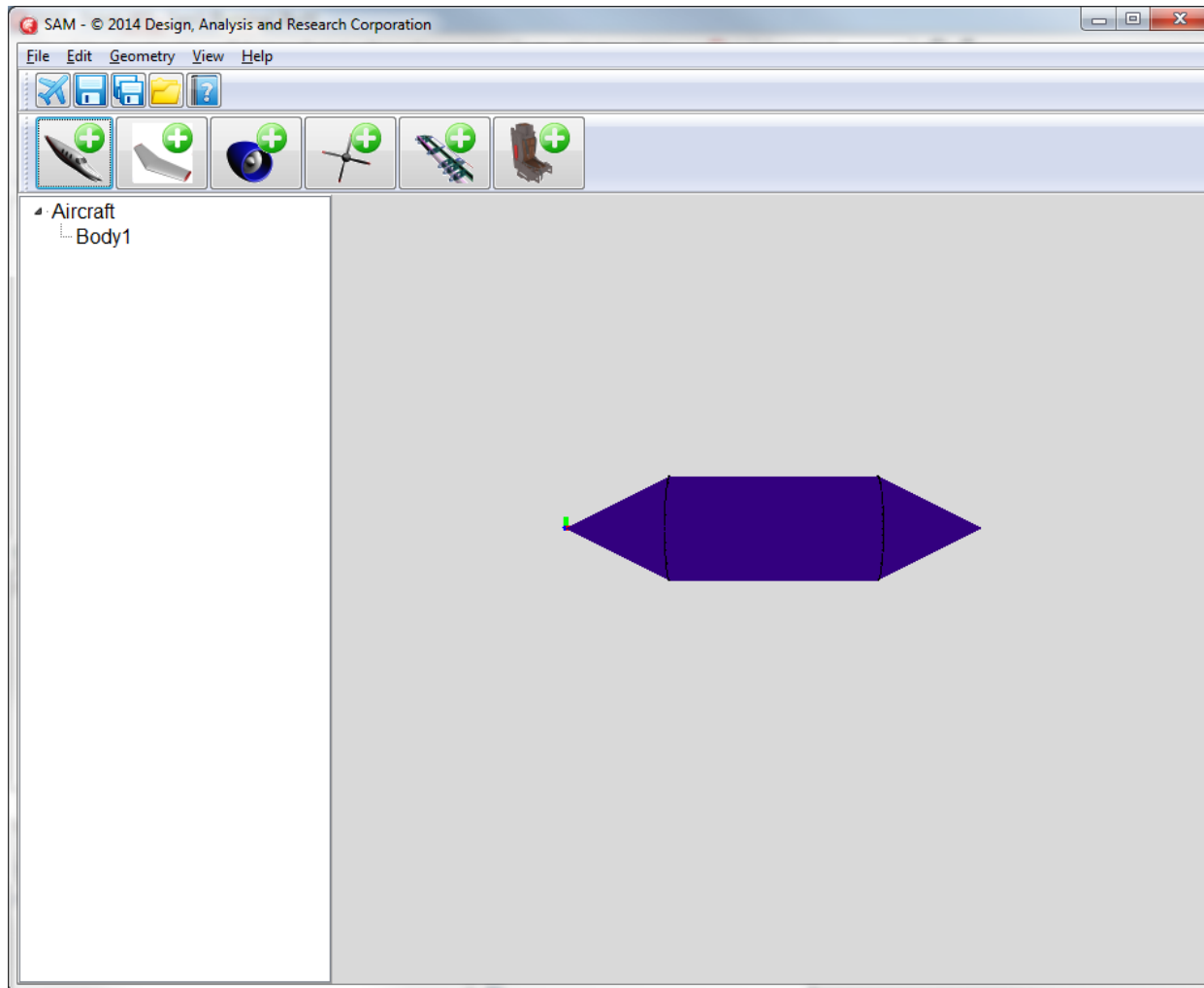
☒ Symmetric ☒ Show Points

[Diagram of a circular cross-section with points and axes]

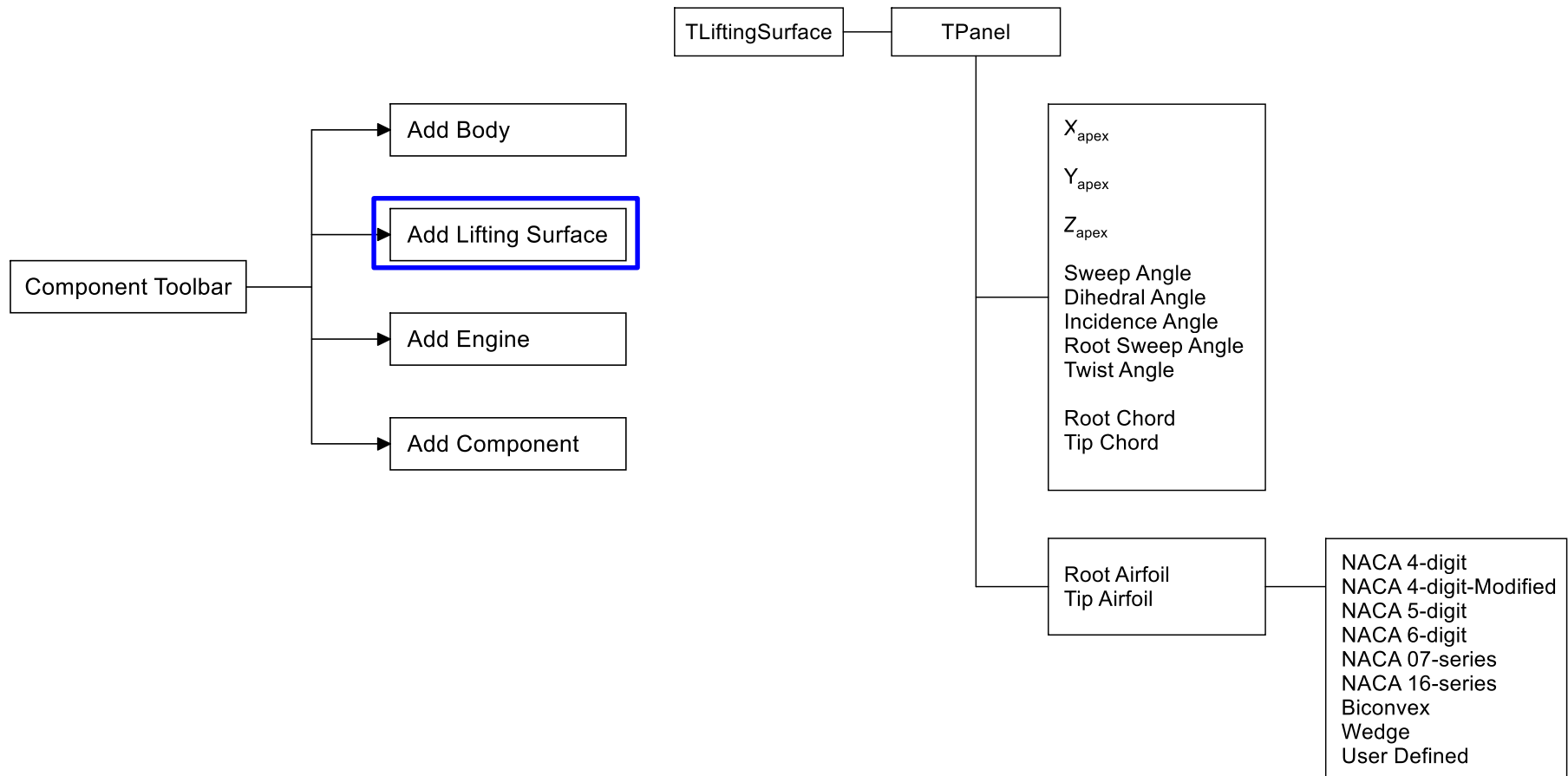
Body Cross Section Generation



Addition of Body to Main Window



Component Toolbar: Lifting Surface Example



Lifting Surface Properties Definition

Wing1


General | Planform | Panel | Airfoil


Name: Wing1


Attached to: ParentComboBox


Show/Hide: ☒ Show ☐ Hide


Visualization: ☐ Wireframe ☒ Surface

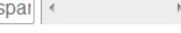
Wireframe Properties: 

Surface Properties: 

Ambient Light Properties: 

Diffuse Light Properties: 

Specular Light Properties: 

Emitted Light Properties: 

Transparency: Transpa

Shininess: Edit1

Wing1

General | Planform | Panel | Airfoil

Position

X_{Apex} : 0

Y_{Apex} : 0

Z_{Apex} : 0

Scale

Scale: 1

Mirror: ☒ None ☐ X-Y ☐ X-Z ☐ Y-Z

Planform Parameters

i_{LS} : 0

Γ_{LS} : 0

N_c : 40

Wing1

General | Planform | Panel | Airfoil

N_{Panel} : 1

Panel Parameters

Panel: 1

N_b : 0

c_r : 5

c_t : 2

X_r : 0

X_t : 0

b_{Pnl} : 10

Γ_{Pnl} : 0

☐ Match Airfoil to Dihedral

ε_t : 0

Root Airfoil: Default

Tip Airfoil: Default

Wing1

General | Planform | Panel | Airfoil

Name: NACA 0012

Airfoil Family: NACA 4-Series

Designation: 0012

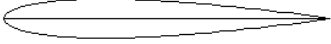
Spacing Bias: Half Cosine Fwd

☐ Show Points

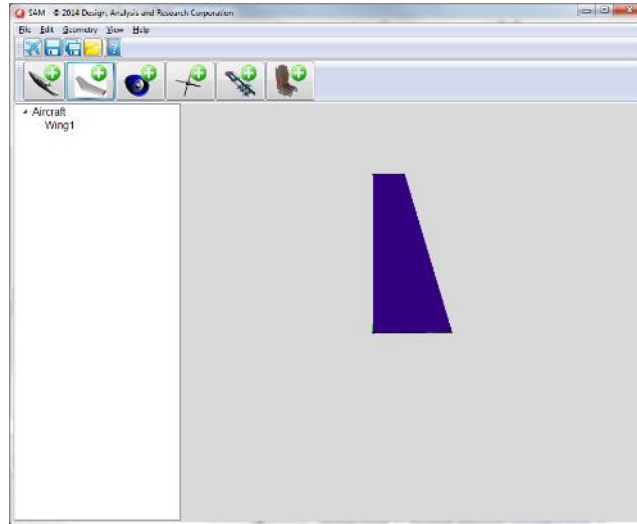
Output

R_{LE}/c : 1.58669298 %

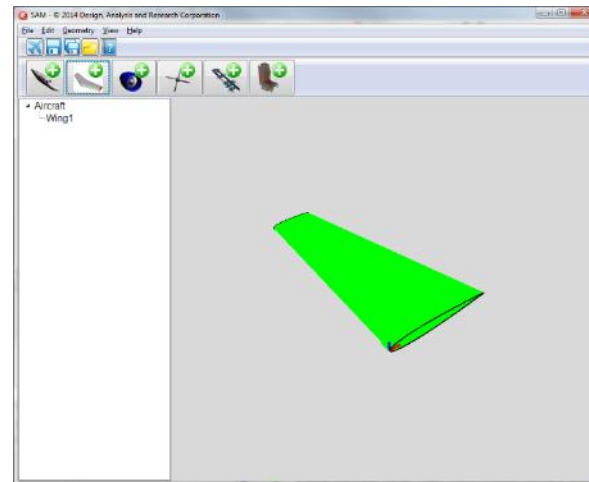
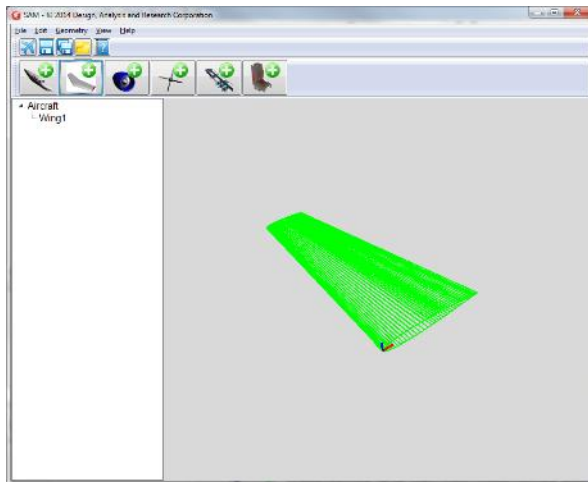
ϕ_{TE}'' : 15.4081774857695 deg



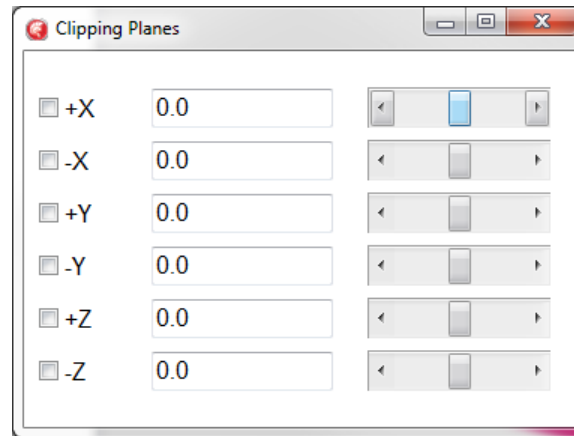
Addition of Lifting Surface to Main Window



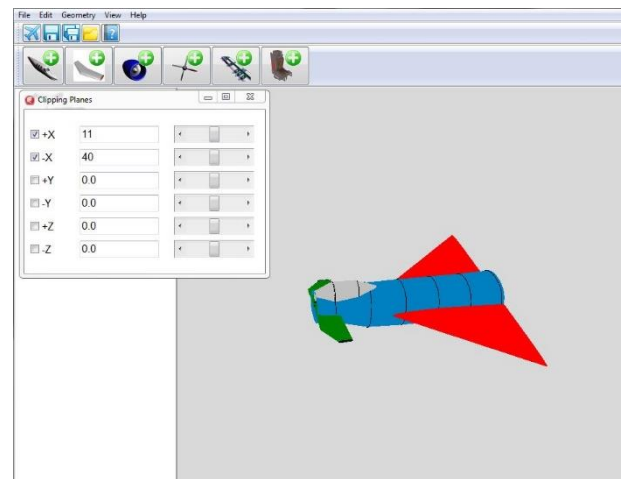
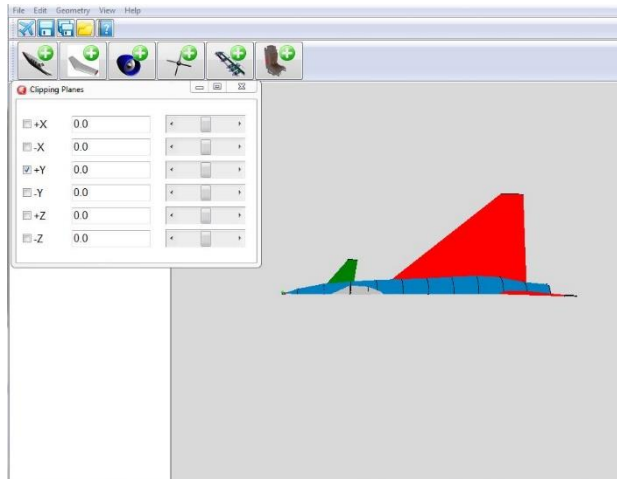
Lifting Surface Wireframe and Surface Skin



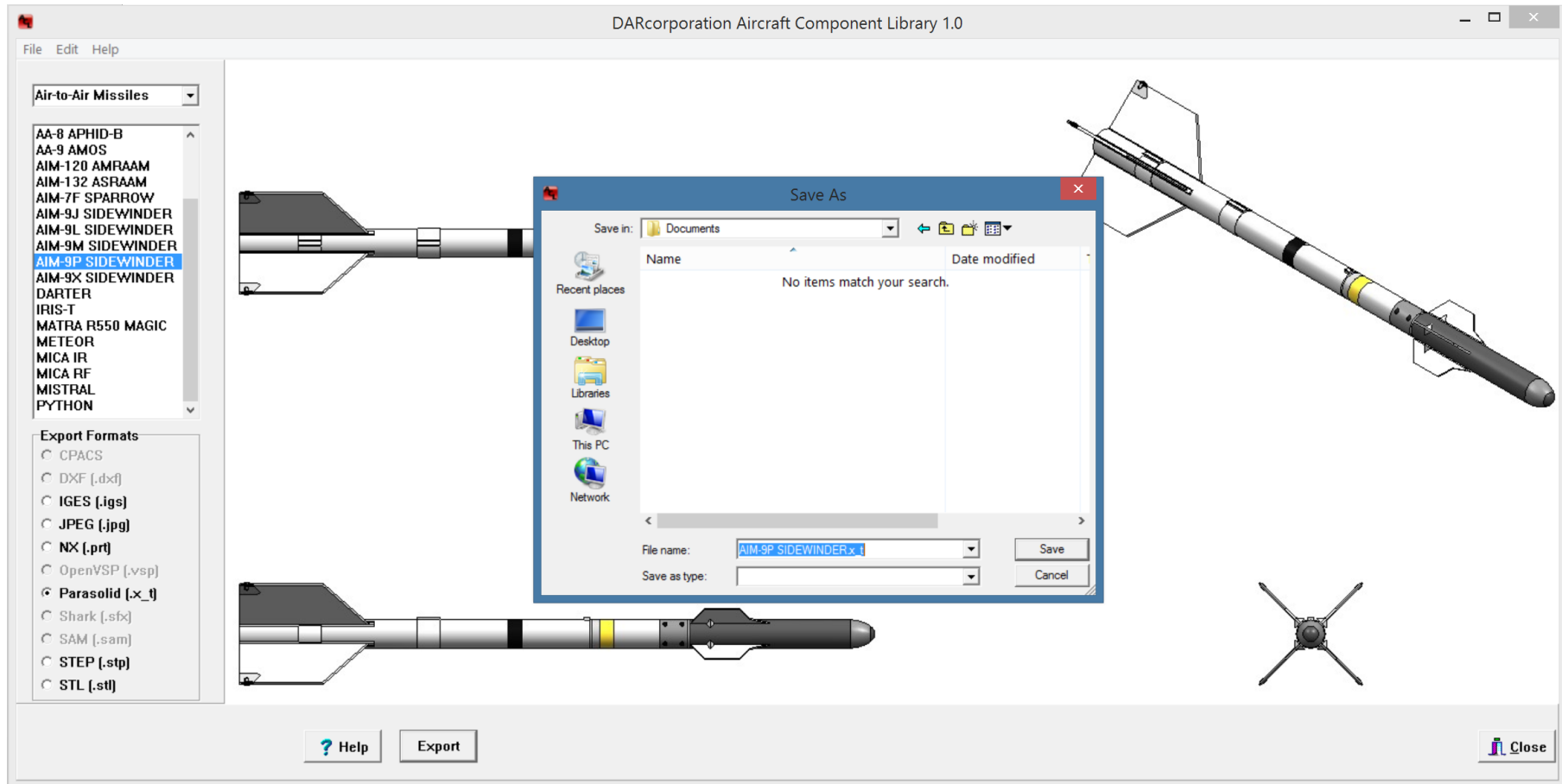
Clipping Plane Selection Window



Clipping Plane Example



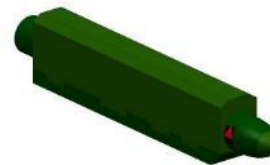
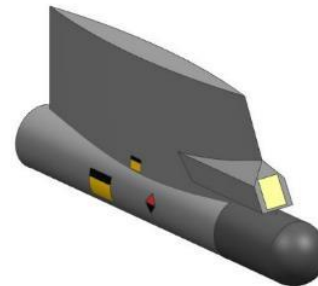
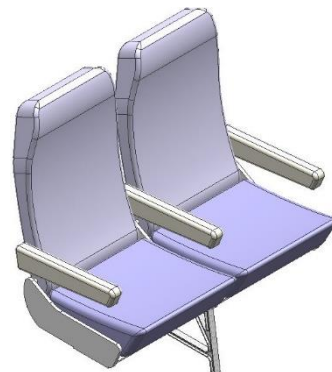
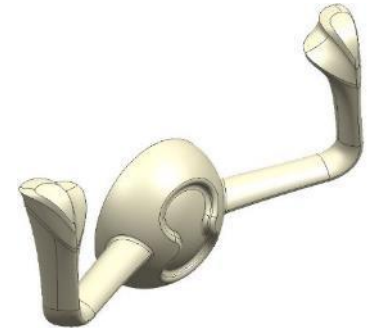
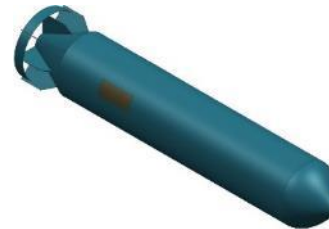
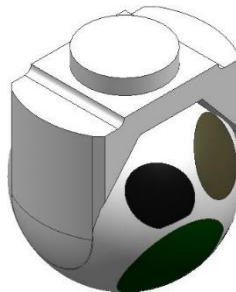
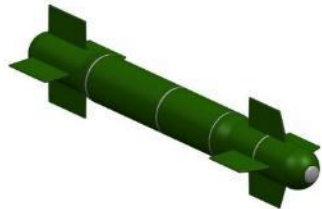
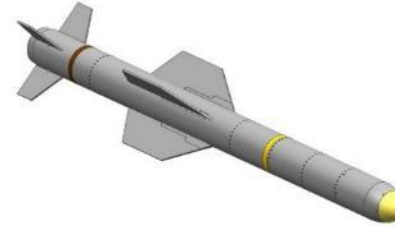
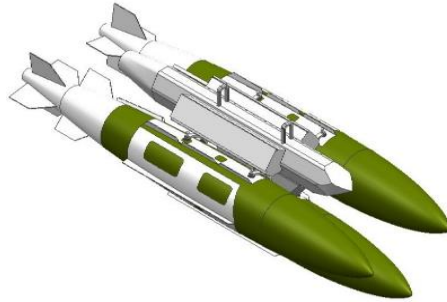
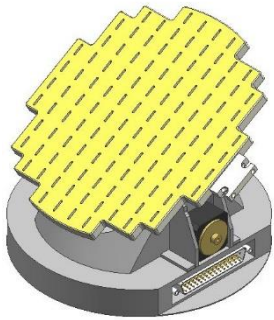
ACL



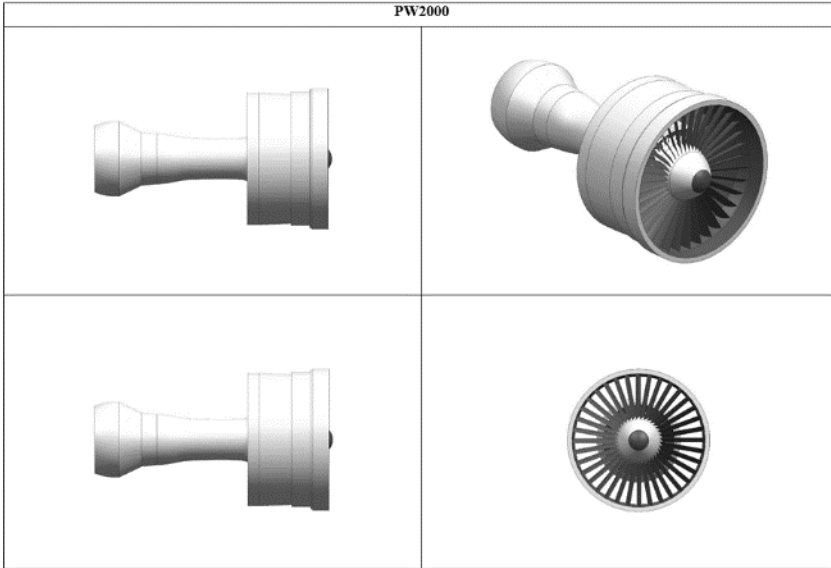
ACL

Name	Number of Models
Aircraft	>100
Air-to-Air Missiles	23
Air-to-Ground Missiles	26
Anti-Ship Missiles	6
Bombs	25
Controls	5
Ejection Seat	1
Engines	7
External Fuel Tanks	5
Gimbal Sensors	4
Gun Pods	5
Guns	9
Helmet	1
Human	1
Mission Pods	10
Radar and Antenna	8
Seats	4
Vehicles	4
Weapon Racks	9

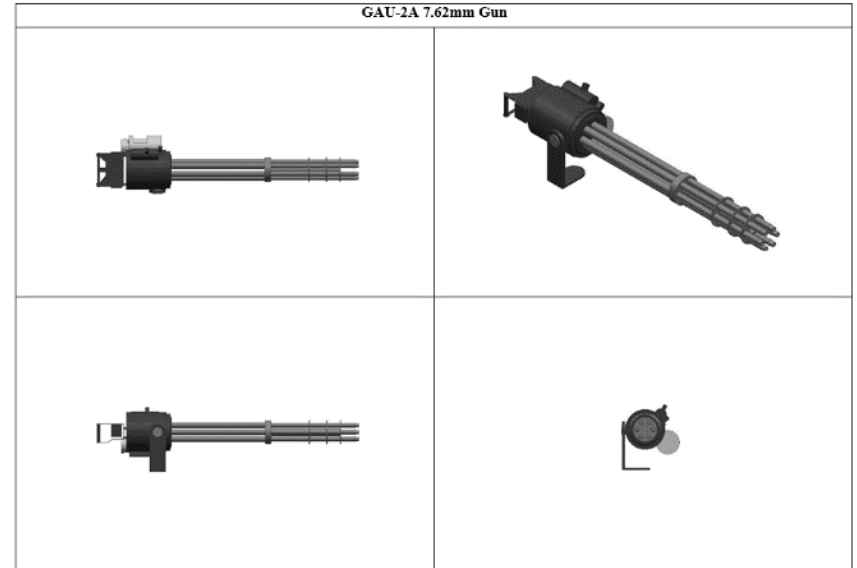
ACL



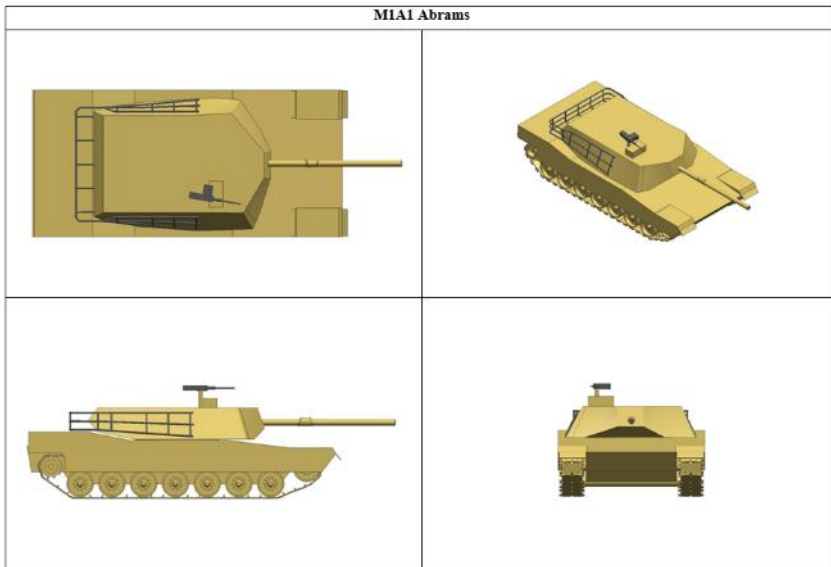
PW2000



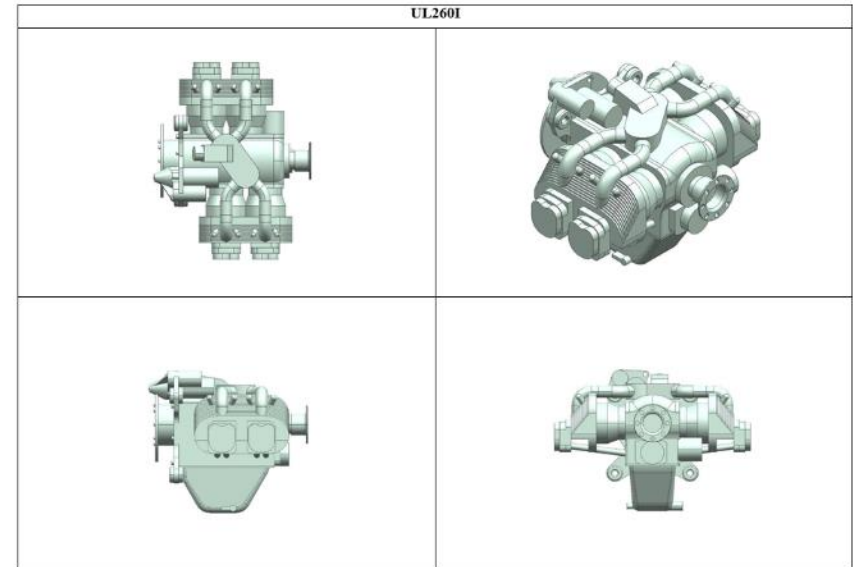
GAU-2A 7.62mm Gun



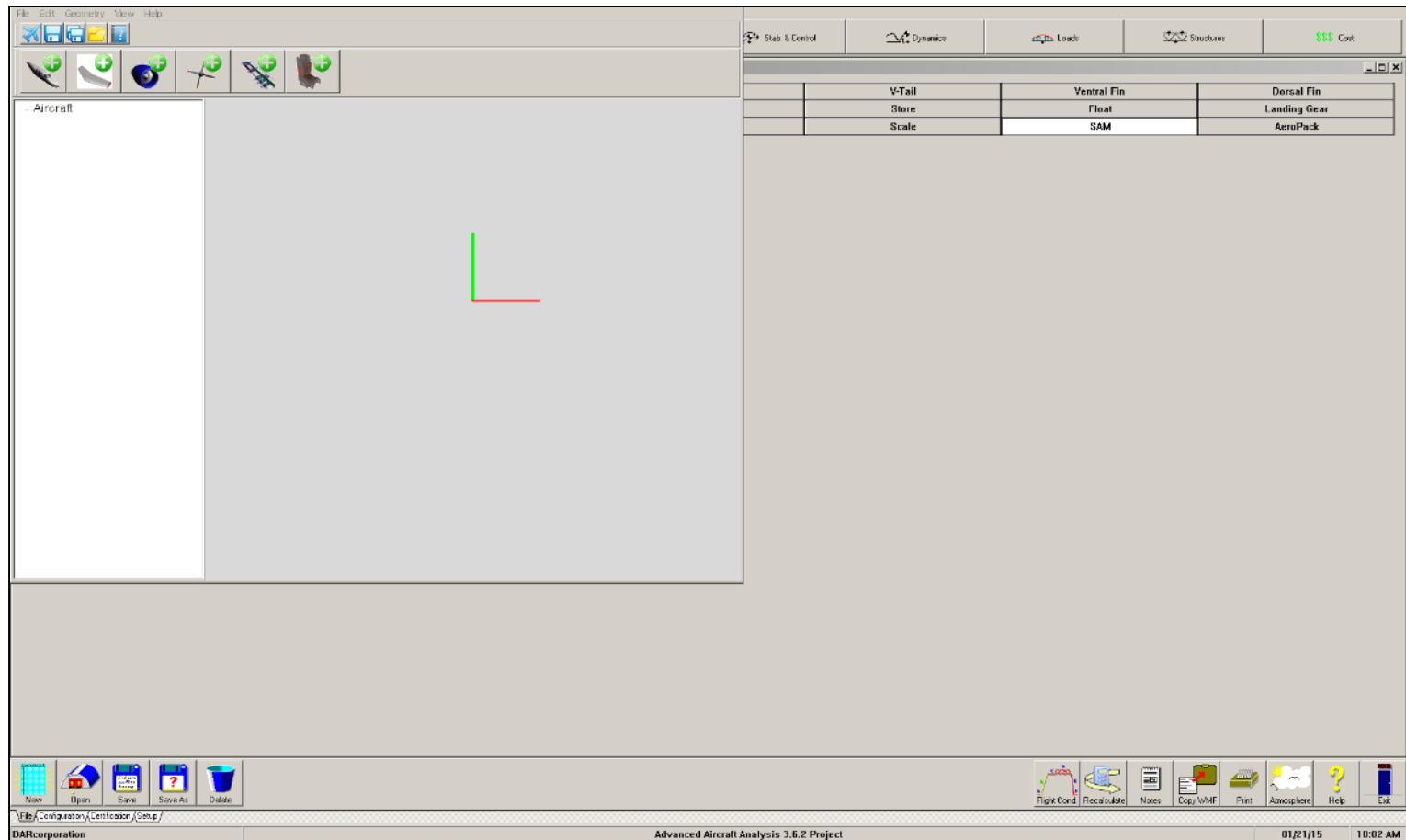
M1A1 Abrams



UL260I



SAM in AAA Framework



ACL in AAA Framework

Weight
 Aerodynamics
 Performance
 Geometry
 Propulsion
 Stab. & Control
 Dynamics

User Defined Weight Center of Gravity: Flight Condition 1

Calculate
 Clear Out
 Import Table
 Export
 Theory
 Close

Input Parameter

Number

User Defined Weight Table

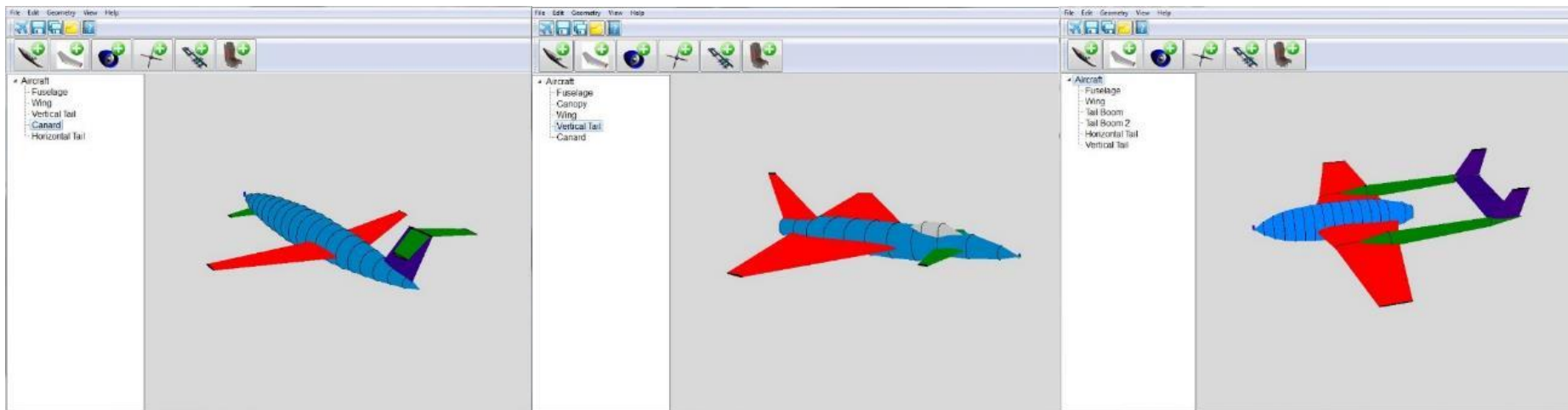
#	Component Name	Weight lb	X _{cg} in	Y _{cg} in	Z _{cg} in
1	AIM-132 ASRAAM				
2	AIM-120 AMRAAM				

Output Parameters

W_{user} lb
 X_{cg_user} in
 Y_{cg_user} in
 Z_{cg_user} in

SAM:

- 3-D Parametric Modeling
- Quick Generation of Novel Layouts
- Inboard Profile
- Backward Compatible with AAA
- Export/Import Options from Shark/AeroPack
- CPACS



- Further Implementation in AAA
- **Smarter** Aircraft Modeler:
 - Increased number of Data Exchange Formats
 - Knowledge Based
 - Improved Input



- Database of Aircraft (> 100 examples)
- Batch Mode for MDO
- Linking to Panel Code/CFD
- Linking to Flight Dynamics (J2Aircraft)
- Airplane Sizing
- SolidWorks/AeroPack
- Siemens NX/AeroPack

Thank You

Questions?

TUESDAY, JANUARY 05, 2016

ACD-01. Aircraft Design Issues I

ACD-02. Aircraft Design Issues II

ACD-03. Aircraft Wing Design

WEDNESDAY, JANUARY 06, 2016

ACD-04. Electric Aircraft Design

ACD-05. Transport Aircraft Design I

ACD-06. Aircraft Design Tools

ACD-07. Transport Aircraft Design II

THURSDAY, JANUARY 07, 2016

ACD-08. Unmanned Aerial Vehicle Design

ACD-09. Conceptual Aircraft Design Working

GNC-30/ACD-10. Aircraft GNC I

ACD-11. Micro Air Vehicle Design

GNC-34/ACD-12. Aircraft GNC II

FRIDAY, JANUARY 08, 2016

ACD-13. Aircraft Design Optimization