

Development of VUT 001 MARABU aircraft



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VUT 001 MARABU = Experimental aircraft to support development of systems for civil UAVs / UAS

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Introduction

Major issues connected with development of UAVs:

 Non-existence of widely accepted regulation requirements for development and operation of civil UAVs

Serious issue affecting development and testing of UAVs in civil airspace. Especially European airspace is "overcrowded" and offers very limited possibilities for UAV development and operations. Use of special testing ranges may be very expensive.

 Simultaneous development of 2 critical elements: Aerial Vehicle and Ground Control Station

Simultaneous development of 2 critical elements makes first flight tests risky.





Project VUT 001 MARABU Project number: FI-IM3/041

Development of civil UAV platform supported by Ministry of Industry and Trade (Czech Republic), held together with industrial partners (2006-2009)

Solution proposed by prof. Pistek

- Development of a "Flying Platform" (at the first stage proposed as piloted aircraft with 600kg *MTOW – to overcome legal issues)*
- Preparation of the experimental aircraft for integration of equipment and systems developed for UAVs (based partially on COTS components) .. and step-by-step integration of suitable UAV systems
- **Development of new propulsion units**

Project Partners

Coordinator:

Letecký ústav (IAE) FSI VUT v Brně, Technická 2, Brno, Responsible person:

Prof. Ing. Antonín Píštěk, CSc.

Partners:

První Brněnská strojírna Velká Bíteš, a.s., Vlkovská 279, 595 12 Velká Bíteš

JIHLAVAN airplanes, s.r.o., Znojemská 824/64, 586 01 Jihlava

PLASTSERVIS-L, s.r.o., Nová Ves 48





Project VUT 001 MARABU

Project definition

Regulation status (EASA, CAA, Traffic rules)

- -Traffic at civil spaceCivil Permit to Fly, Civil standards,CAA approval
- -UAV control equipment and application presumption, Equipment volumeequivalent to pilot mass and volume
- -Flight learning ...step by step 1 pilot and equipment ...Payload 180 kgs

Power plant

- -Low and high altitude flights
- -Long endurance Low consumptionpiston engine
- -Testing of JET power

Future exploitation of aircraft for sport flying

- Possibility normal using of plane (2 pilots)





Perspective Development of VUT 001 Marabu

Industry interested in UAVs exists in Czech Republic

Association of the Aviation Manufacturers (UAV Group)

Czech Unmanned Systems Manufacturers Association

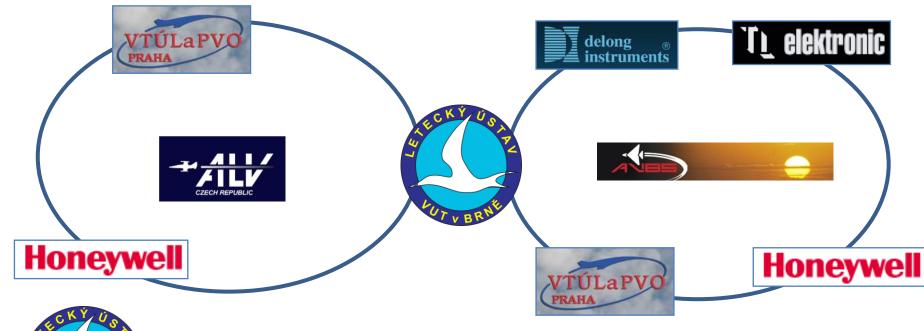






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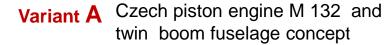


HERTI (developed in Warton, UK).





Early Development



Variant B

Two TJ100A jets on rear part of the fuselage

Variant C

Rotax piston engine and butterfly tail unit



History of VUT 001 Marabu concepts

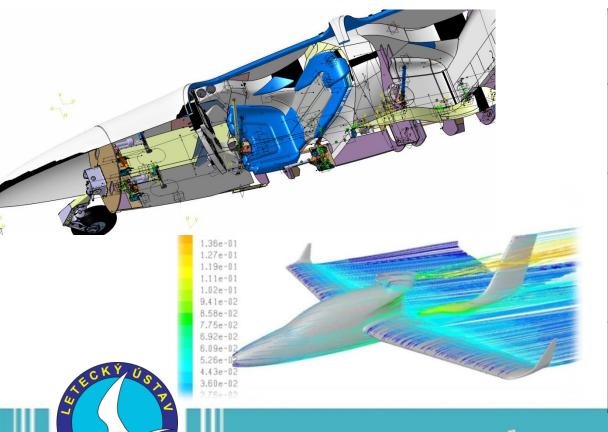








- CAD approach was used from conceptual design through preliminary design up to detail digital model of the prototype.
- Aerodynamic concept was optimized using CFD methods to enable excellent performance characteristics.



VUT 001 Marabu characteristics				
	Length	8,1 m		
	Height	2,4 m		
Weights	Max. take-	600 kg		
	off			
	Empty	380 kg		
	Max. fuel	141 kg (188		
		liters)		
Performa	Max.	260 km/h		
nce	speed	200 KIII/II		
	Endurance	7 h		



- Structure of the fuselage was designed using composite materials to enable light and stiff structure.
- Modern FEM for structural analysis were applied to further reduce weight of the structure and to enable quick definition of dimensions for critical structural parts.

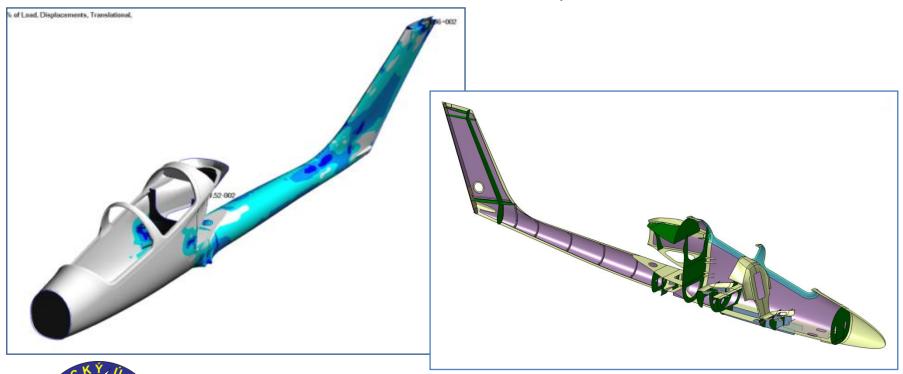






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Major requirements on VUT 001 MARABU systems:

Movement to More-Electric-Aircraft (MEA) concept:

Target is to make as many systems as possible electrically driven. This will enable smooth step-by-step transition from fully piloted aircraft (today) into fully automatic aircraft (in the future).

Step-by-step integration of UAV systems should be enabled:

Based on the statistical analysis, most UAVs are designed with max. take-off weight of 600kg or less (88% of designed UAVs).

Proposed experimental aircraft provides enough space and typical electric system for integration of majority of developed equipment.

Provisions for back-up of critical systems:

Provisions to optionally attach second alternator were made to provide backup function and to increase an overall capacity of the electric system for future experimental applications.





VUT 001 Marabu systems

Comparison of systems in typical conventional aircraft and VUT 001 Marabu

	Type of System		
System	Conventional Aircraft	VUT 001 MARABU	
	(FAR-LSA, CS-VLA)	VOI OUI MARABO	
Primary Flight Controls	Mechanical (push/pull rods,	Mechanical (push/pull rods,	
	cables, etc.)	cables, etc.) – provisions are done to	
		mount autopilot servos	
Trim System	Mechanical (cables, etc.)	Electrical (elevator, rudder)	
Flaps Extension/Retraction	Mechanical (push/pull rods)	Electrical (electromechanical	
		strut)	
Electric System	Simple with 1 alternator and 1	2 alternators and 2 batteries	
	battery (optionally, second alternator	create redundant system with two	
	to increase capacity is used	independent channels.	
	interconnected with the single battery	Additional independent channels can	
	used also for ALT1)	be supplied by own energy sources	
		(batteries, fuel cell stacks, etc.).	







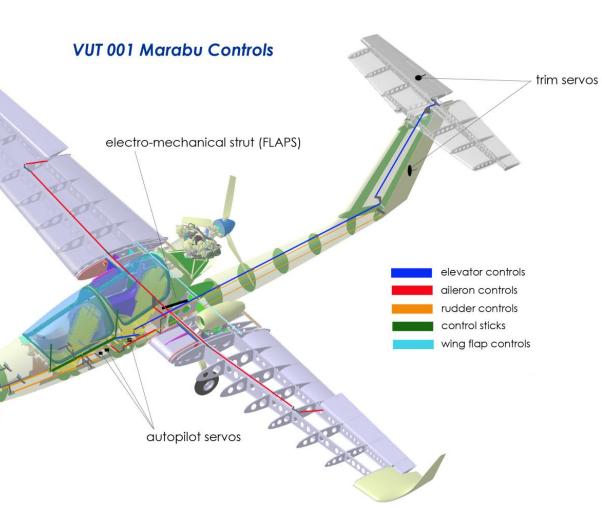






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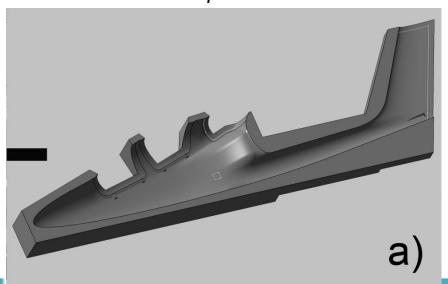




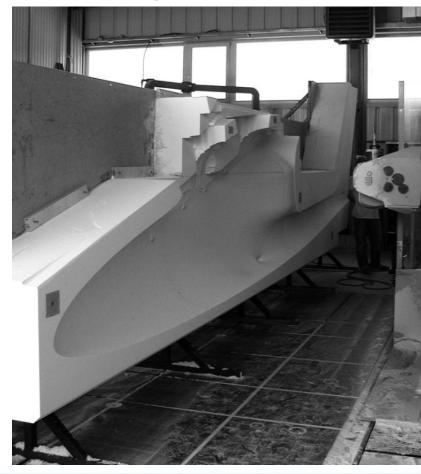
Fuselage production

Specific at academic environment

All activities were in large extend performed by young engineers, researchers and students at the university. This could be done as a result of IAE's long time activities focused on building of capacities for research and development.



Negative mould











First positive model

Final bonding of prototype









First engine test









Required structure tests:

- -Wing tests Ultimate load
- -Centroplane test Ultimate load
- -Test of Ailerons Ultimate load
- -Test of Flaps Ultimate load
- -Winglet test Ultimate load
- -Fuel tank test Ultimate load
- -Fuselage test Ultimate load
- -Stabilizer and elevator tests -Ultimate load
- -Seats and seat belts joint tests -Ultimate load
- -Engine mount test Ultimate load
- -Control system Limit load
- -Nose landing gear drop tests
- -Main landing gear drop tests



Centroplane torion test





Required structure tests:

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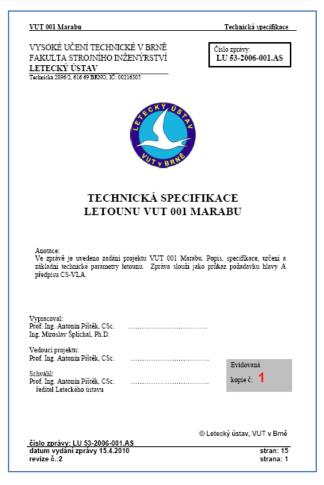
Final prototype assembly



Permit to Flight

Requirements of Czech CAA (before first take-off) •Ground tests:

- · Mass and centre of gravity definition
- · Airplane leveling, real control surfaces definition
- ·Test of pitot-static system
- Engines test + static thrust measurement
- Board instruments calibration
- •Test of electric system
- Test of fuel system
- Test of flaps operation
- Taxing capability and stability
- Flight manual
- · Maitenance manual
- Instruction for test flights







Permit to Flight

ÚŘAD PRO CIVILNÍ LETECTVÍ CIVIL AVIATION AUTHORITY

ČESKÁ REPUBLIKA



CZECH REPUBLIC

Výrobní číslo

ZVLÁŠTNÍ OSVĚDČENÍ LETOVÉ ZPŮSOBILOSTI

SPECIAL CERTIFICATE OF AIRWORTHINESS

č. / No: ZOLZ - 5530

. Výrobce a typ letadla

Mark	and Registration	Designation of Aircraft	Alleran Schar No.
ок	-VUT	Vysoké učení technické v Brně VUT 001 Marabu	001
Kategorie Categories	Experimentá	lní - Experimental	the contract of the contract o
		odatek C – Operational limitation according koncepcí za podmínek dodržení om	

Omezení - Limitatione

podmínek dodržení omezení 1.2.a) až f)

I. Poznávací značka

- Pilot musí být držitelem kvalifikace "zkušební pilot bez omezení", dle AIC C 41/94

 Zkušební lety dle článku 1.1.a) a 1.1.b) L6/II dodatek C musí být prováděny za dodržení podmínek stanovených v SOUHLASU S PROVEDENÍM ZKUŠEBNÍHO

1.1.b) lety k průkazu požadavků předpisu způsobilosti na funkci a spolehlivost za

- LETU (série letů) č.j. 4176-09-401 a pouze v rozsahu programu schváleného Úřadem.
- Pilot musí být vždy vybaven záchranným padákem
- Pilot musí informovat v souladu s 2.b) L6/II dodatek C

Toto letadlo nesmí být provozováno nad územím jiného státu bez povolení jeho Leteckého úřadu.

This aircraft shall not be operated over any other country without Civil Aviation Authority permission of that country.

30-11-2009

Datum vydání - Date of issue (dd-mm-rrrr) - (dd-mm-yyyy)



(Michal Štěpán)

Podpis -Signature



Úřad pro civilní letectví České Republiky sekce technická

Vydáno pod č.j.: 4541-10-401



ROZHODNUTÍ

Podle us anovení § 15 odst. 2 zákona o civilním letectví č. 49/1997 Sb.v platném znění a na základě žádosti vydává ÚCL

Souhlas s provedením zkušebního letu

Společnosti : Vy3

 Vysoké učení technické v Brně Fakulta strojního inženýrství se sídlem : Technická 2896/2

616 69 Brno

Držiteli oprávnění: výrobci letadla

Letecký ústav

který provede s letadlem typu: VUT 001 MARABU

poznávací značka : OK- VUT

Omezení a podmínky pro provádění zkušebních letů :

- Nedilnou přilohou tohoto Rozhodnutí je Instrukce pro provádění zkušebních letů letounu VÜT 001 Marabu č.LU47-2009-001.SM, schválená Úřadem.
- 2. Lety (série letů) budou prováděny z letiště Kunovice LKKU a Jihlava LKJI jako zk. kšební lety pro ověření způsobilosti individuálně vyrobeného letadla dle ustanovení \$9 zákon a č.49/1997 Sb. o civilním letectví v platném znění a budou vedeny jako zk. zšební lety dle předpisu L6/II Dodatek C, bod 1.1 a) a b). Součástí zkoušek letounu bude ověřevání proudového motoru TJ 100C a vlastností letadla s pohonem touto iecnotkou.
- Lety budou prováděny v rozsahu provozních omezení stanovených letovou příručkou a Úřadem schválenými programy zkoušek.
- Zkušební lety budou prováděny v prostoru vymezeném letištním řádem letiště na němž budou prováděny, v rozsahu a s omezeními dle schváleného programu zkoušek.
- Lety mohou být provedeny pouze se souhlasem provozovatele daného letiště.
- Ošetřování, přípravu a zajištění letadla k letu budou provádět osoby dle seznamu pozemního personálu uvedené v Instrukci pro zkušební lety.
- Žkašební lety budou provádět piloti s kvalifikací pro zkušební lety bez omezení dle AIC C 41/94, uvedení v Instrukci pro zkušební lety.
- Na zkušební let vydá příkaz k letu odpovědná osoba provozovatele VUT Brno.
- Za zpracování PRG a MET a vyhodnocování zkoušek pro letoun je odpovědná skupina aerodynamiky, pro motor TJ 100C pracovníci PBS Velká Biteš.
- Platnost tohoto Souhlasu je omezena na odlětání schváleného programu zkoušek, nejpozději však do 30.11,2011.

UN CLASTIONOSO



First Flight



First test flights were performed in Kunovice (south east part of the Czech Republic).



Test-pilot Stanislav Sklenar reported excellent handling and performance characteristics.



Experimental airplane VUT 001 Marabu was recently awarded with Gold Medal MSV 2010.

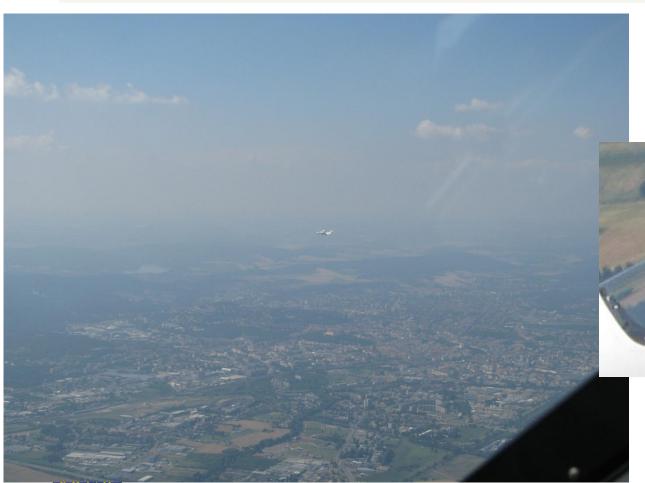
MSV 2010 = the largest presentation of industrial technologies in the area of Central Europe.







First cross-country fly



Fly from LKKU to LKJI for next test flights

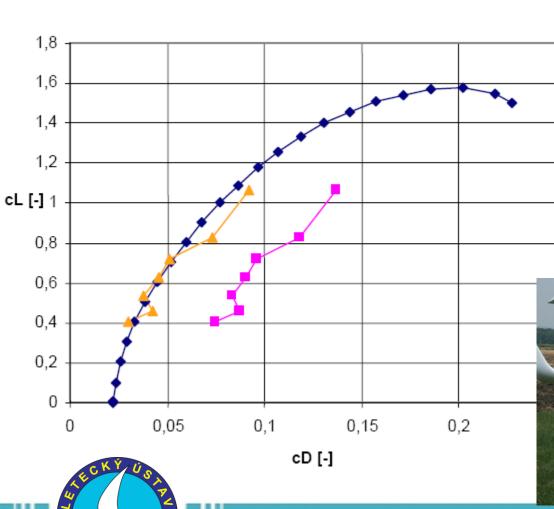


EXPERIMENTAL





Flight Tests



Aerodynamic polar – significant influence of fuselage-wing interference drag







Flight Tests







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Perspective Development of VUT 001 Marabu

Applications and perspective development of VUT 001 Marabu

A) UAV activities

- Finalization of full UAV version automatic control system
- Development of sensors and components for critical systems of **UAVs**
- Low cost testing of equipment for various missions
- Simulation of different mission profiles

B) Non-UAV applications

- Flight measurements of characteristics for developed jet engines
- Development and production of VUT 061 Turbo modification for turboprop engine TP-100 (180kW)
- Development and production of VUT 051 Ray modification for **non-conventional propulsion** (electric engine, on-going activity)



Pilot Stanislav Sklenar and prof. Pistek





Experimental jet engine TJ100M

Small jet engine developed in PBS, Velká Bíteš.

TJ 100M



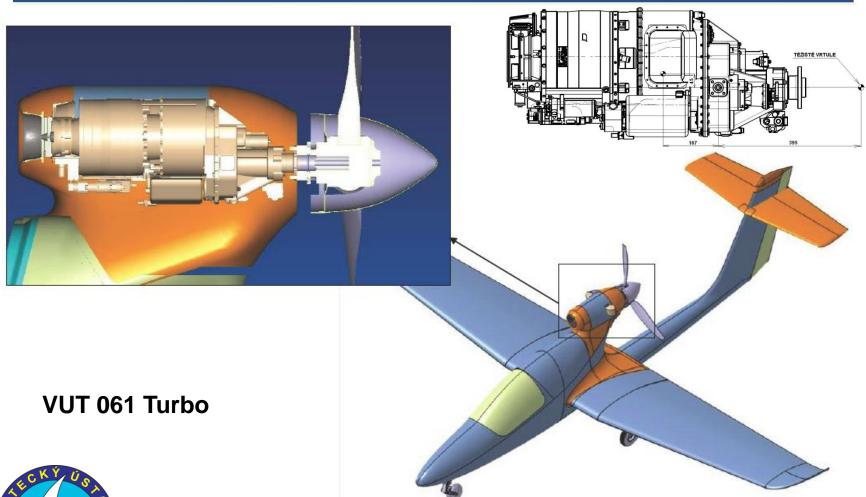


TJ100M jet engine with thrust up to 1100N designed for UAV applications

Yabhoon aerial target (uses TJ100)



Perspective Development of VUT 001 Marabu







Thank you for your attention ...

Acknowledgment

Some of the presented activities were supported by Ministry of Industry and Trade (in the frame of grant project FI-IM3/041).











Detail information will be published on http://lu.fme.vutbr.cz













We are looking for aerodynamics, structure experts

Detail information are published on

http://www.cedesa.eu



