Presented by

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- The Airbus A380 Towards a New Future for Air Transport



The future has arrived



A380 Family

159 firm orders16 customers

4380

(end April 2006)

111111

A38

21st Century flagship



560t 555 seats 8000 nm EIS 2006

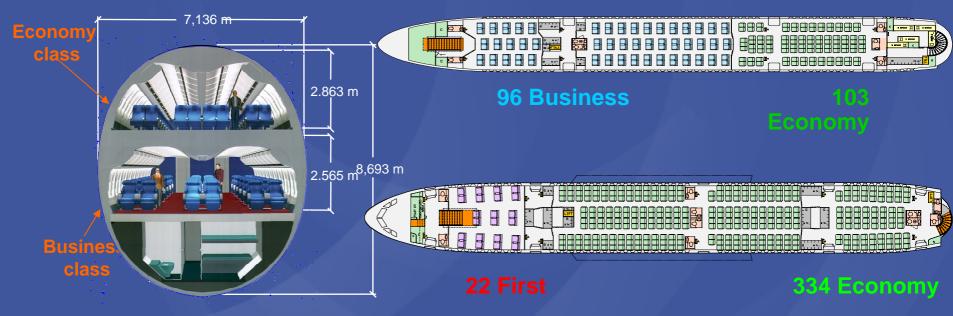
590t 150 t 5620 nm EIS 2008



Page 4

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A380 Cabin layout



- 2 full decks
- 4 aisles



A360 main deck, Economy Class sealing configuration



Providing more capacity, more comfort



Upper deck - true widebody cabin 2-2-2 business class, common product with Airbus long range family





Large innovation potential





The lowest cost by far

Advanced systems & manufacturing processes Advanced materials

A380

WON



Advanced powerplant



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A380 in 300 seconds





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A380 maintenance advantages

Integrated Modular Avionics

- Common hardware
- Software upgrades onboard
- Flexibility for reconfiguration

Structures

- GLARE and CRFP for less fatigue and corrosion
- Laser Beam Welding
- Standard repair procedures

Onboard Maintenance

..........

- Cockpit server for all Manuals
- E-logbook, Airman, PMATs

Fly-By-Wire

- New Electro-Hydraulic Actuation
- Superior redundancy for High dispatch reliability

Maintenance Programme

- Longer check intervals
- Low scheduled requirements

Electrical system

- Variable frequency
- New Interactive Maintenance

New Hydraulics

- Only 2 circuits, 5000 psi
- Maintenance free accumulators

Leading to 25% lower DMC per seat than the 747-400



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The **GREEN GIANT** for our planet

mannet

VANAAAA

The first long-haul aircraft with less than 3 litres per pax/100km fuel consumption*

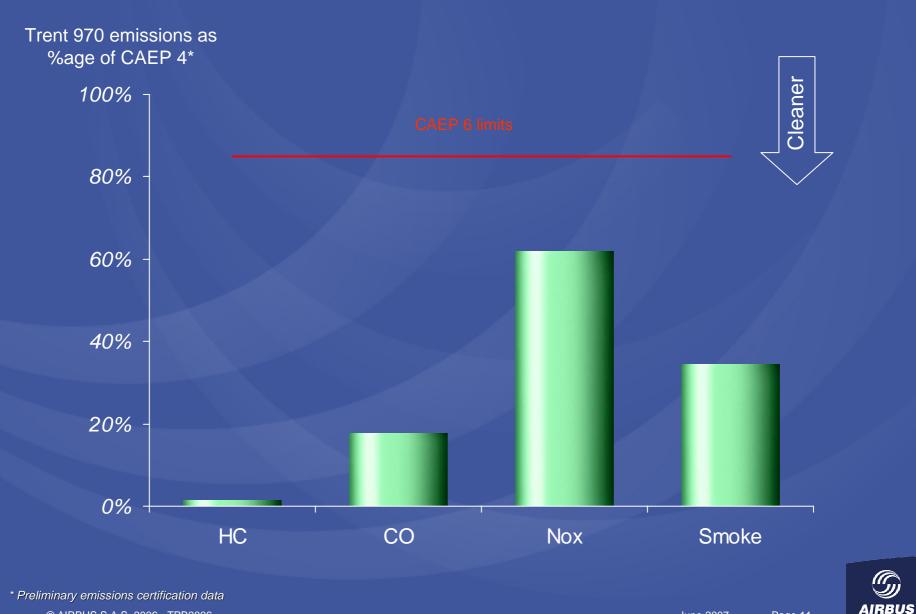
Over 80 miles per pax per USG



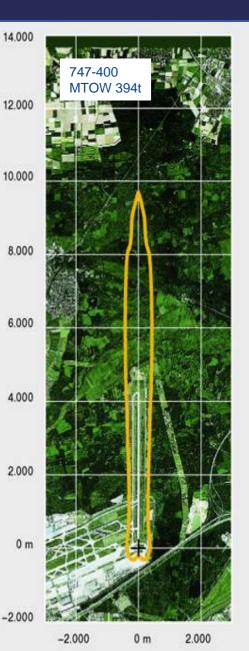
* 5000 nm sector, Typical International Flight Profile, 555 pax © AIRBUS S.A.S. 2006 - TPB2006

June 2007 Page 10

A380 and the environment



A380 vs. 747-400 noise contour





85dB(A) Noise Contour for take-off at FRA as calculated by Lufthansa with input of Boeing and Airbus nominal noise data for same take-off conditions

The A380's double advantage

The technology effect: half the noise per movement



2 *The capacity effect:* fewer movements



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A380 likely routes



Subject to change based on current A380 customer and airport statements - Status as at end of 2005

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Page 13

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2010 airport readiness status

By end 2010, 69 airports will be ready for the A380



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A380 airport compatibility is proven

Large airports, for operational checks



Small airports, for performance testing



Medellin, Colombia



Pointe-à-Pitre, Guadeloupe





Tarbes, Southern France



Iqaluit, Canada

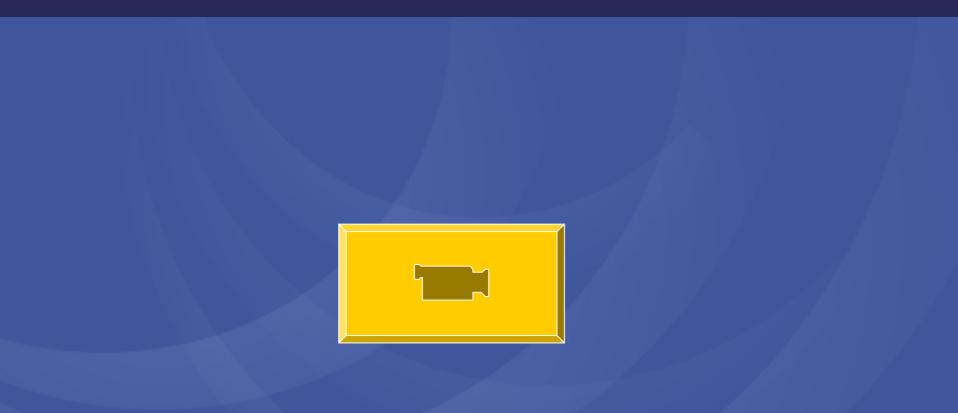
More than 20 airports already visited





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A380 First flight





A380 Flight tests

• From 27th April 2005 to 8th June 2006

- Aircraft flying : MSN 1, 4, 2, 7, 3
- 430 flights
- 1416 flight hours
- More than 100 pilots, including some 50 airline pilots



A380 Flight tests

• <u>Test objectives</u>

- Check of the aircraft behaviour
 - Direct and normal flight control law
 - Effect of speed, Mach number, altitude, aircraft weight and centre of gravity position
 - All flight phases, from take-off to landing.
- Check of the adequate functioning of the various aircraft systems during the various flight phases.

• <u>Main results</u>

- Excellent aircraft behaviour
- Control laws, and auto flight already very mature
 - -Aircraft behaviour close to simulator
 - -Successful autoland on flight 17, 35 days only after first flight
- Major systems working as intended during normal operations.
- Early safety checks carried out flawlessly
 - -Landing gear gravity extension
 - -Ram air turbine extension and functioning
- ■Engine relight © AIRBUS S.A.S. 2006 - TPB2006



Water trough tests

• 1 acceleration + 1 deceleration at ~70 kt





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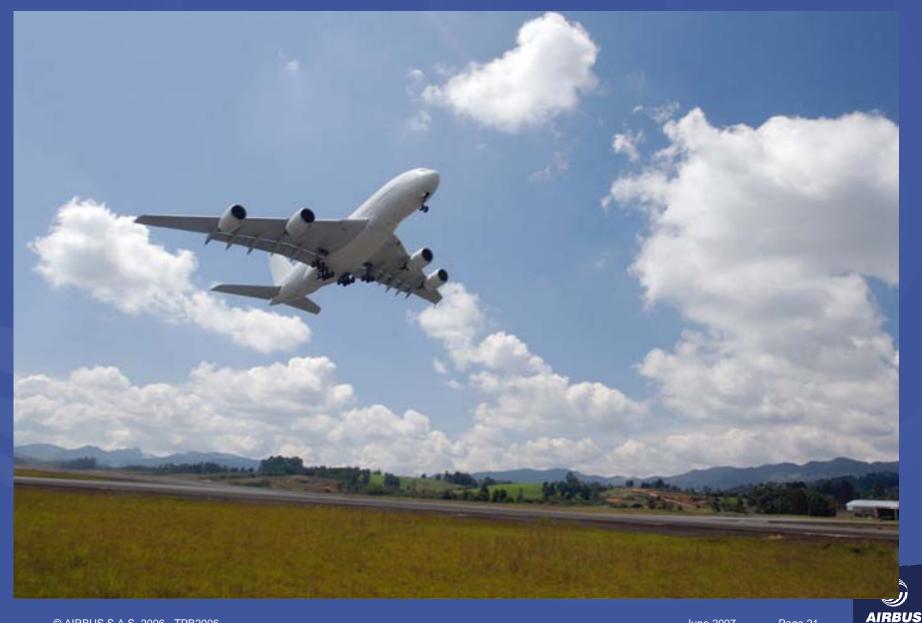
Frankfurt Airport compatibility checks – 29th October 05



Page 20

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Hot & high campaign in Medellin (Colombia)



Cold weather campaign in Iqaluit (Canada)









Cabin evacuation test – 26th March 06

- Certification requirement : Evacuation must be fulfilled in less than 90 seconds using half the number of doors
- Result : 873 people evacuated in ~80 seconds



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Video A380 Evacuation Test





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Cabin Virtual First Flight – 10th May 2006

• <u>TEST OBJECTIVES :</u>

- Test all cabin functions on ground with a representative passenger and crew loading during 5 hours on MSN 2.
- 474 passengers
- 22 crew members





Major R&T impacts from Flight Physics

High Reynolds Number W/T Testing

Reduction scatter in the performance figures through the development and application of High Re Test techniques: better prediction and therefore less margins for Performance guarantee (Conventional Tunnel = +/-0.75 % vs +/- 0.25 % in Cryogenic Tunnel) – To be translated into **1.3 dc (0.5 %) drag benefit**

Nacelle anti-ice Cyclone concept

50% reduction in Nacelle Anti-Icing system weight (**100kg**) plus reduced in Anti-Ice System **Maintenance**

Advanced Load Control

Reduction of 2200kg of wing weight through further improvements in Aircraft load control (fatigue/manoeuvre/ turbulence Wing Loads alleviation)

Integrated Wing Design

- VHBR Engine Integration
- High Speed Wing Design
- Advanced CFD Simulation

More In-board loaded wing gave **weight reduction** of **4000kg** (for a slight increase in induced drag) through improved understanding of High Reynolds/Mach wing aerodynamics plus a **drag reduction** of **3-4 dc** (1.5%) through improved wave & installation drag handling.

Methodology for Wake Vortex Prediction

Validated methodology for wake vortex prediction and enabling A380 classified in the same category as B747 (instead of super heavy: +2NM) Benefits for marketing & **Airport capacity**.

Optimal Tail and Empennage

Variable thickness distribution of VTP/HTP gave mass reduction of **350kg**, improve tail flow saved approximately **1.5dc (0.5% of drag)**.

NAMES OF TAXABLE

Droop Nose

Lower drag, improved lift / drag ratio for take-off performance, tailored maximum lift.

Major R&T impacts from Powerplant

Inlet acoustic liner (0-splice)

Automated (FMS) Noise Abatement Departure Procedure

S

1 st

Noise reduction landing gear fairing

> Nacelle anti-ice Cyclone concept

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Wing over-pressure tube noise suppression



Major R&T impacts from Structure



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June 2007 Pag

Page 28

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Major R&T impacts from Systems

SI

Dual air conditioning pack

New concept based on two redundant pack

Integrated and modular avionics (IMA)

AFDX high speed bus network

Cockpit

Interactive man /machine interface

Electrical generation

Solid state power controllers

Variable frequency power generation

Vehicle flight

Control Enhanced vehicle control laws e.g. "Brake to Vacate"

Electro-hydraulic actuators

Two hydraulic (5000 psi) + two electrical channel architecture

On board maintenance systems

Based on open world network



A380 – first feedback



An excellent aircraft, already well appreciated by pilots and which will be very soon appreciated by passengers

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The A380: flying today...

... an all-new, 21st century design

... on track in payload and range - and making much less noise

... achieved JAR and FAR certification - to the latest standards

... with true development potential - a future-proof family

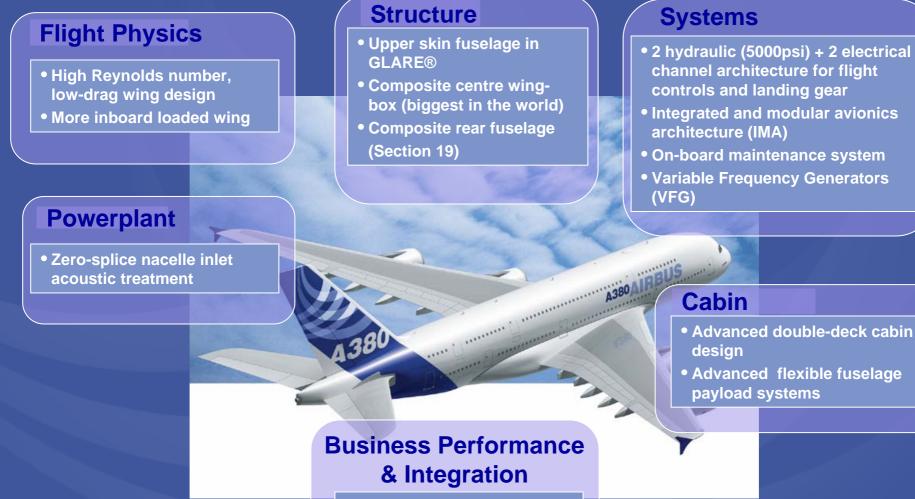
... it will be the flagship of the 21st century !







Major R&T impact on A380



- Overall aircraft configuration
- Concurrent Engineering
- KBE for wing design

Some misperception...

- Correctly pointing out A380's lower fuel / seat (than any 747)
- The figures on A380 emissions* are, however, as follows:



A380 vs 747-400 (RR engines)

HC	-96%
CO	-59%
Nox	-25%
smoke	- 9%

* Preliminary emissions certification data from RR

A380-800 / RR Trent 970 vs 747-400 / RR RB211-524H-T



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