TP400-D6 Turboprop A European Collaboration Programme

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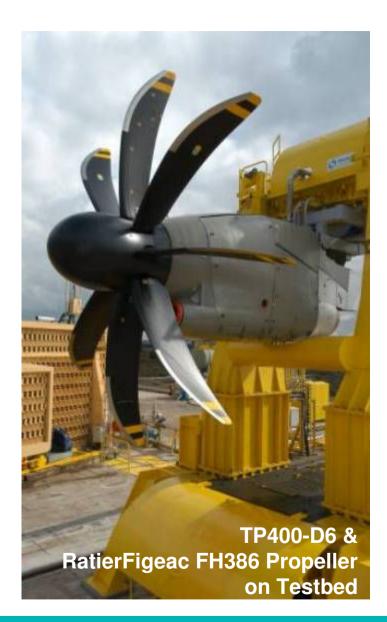


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Why a Turboprop ...?

- 1 Advantage in fuel efficiency for a given payload/range target
- **Particular operational capabilities**
 - 1 Steep Descent
 - 1 Ground Manoeuvrability
- 1 Thrust responsiveness during low-level flight missions
 - 1 Airdrop missions
 - **Field Performance**



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The Engine Team



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Europrop International GmbH





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Heritage of European Co-operation

Tyne for C-160 Transall

RR / Snecma / MTU / Techspace Aero



EJ200 for Eurofighter

RR / MTU / ITP / AVIO



Adour for Jaguar

RR / Turbomeca



Olympus for Concorde

RR / Snecma



RB199 for Tornado

RR / MTU / Avio



Larzac for Alphajet Snecma / MTU / Turbomeca / KHD



MTR390 for Tiger MTU / Turbomeca / RR



RTM322 for NH90 RR / Turbomeca



We are doing this not for the first time... but still it is "more commercial now"

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TP400-D6 Partner Company Workshare



Performance Whole Engine Model Air/Oil System Intermediate Casing HP Compressor Internal G/B LP Shaft Hot Strut Sensors & Probes

Flying Test Bed Vulnerability Assessment Type Certificate Holder Validation / Certification Project Management Integration Management Installation Management Interface Control Prop G/B Management Configuration Control



Development Tests Test Equipment

Front Structure Exhaust Case LP Turbine Dressings



Installation Control System Development Tests HP Turbine Combustor Acc G/B



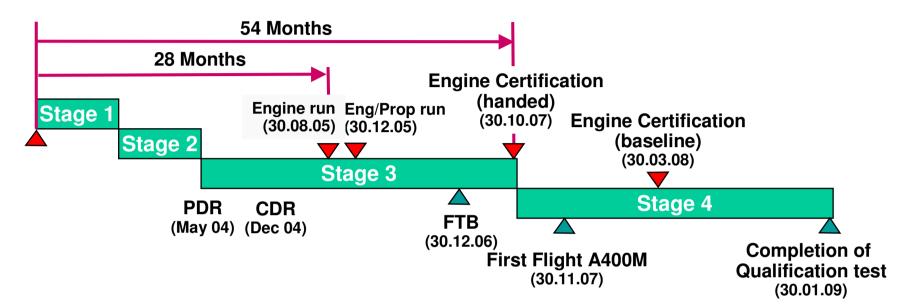
Production Assembly Control System Development Tests

IP Compressor IP Shaft IP Turbine

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TP400-D6 – Engine master programme



- n Fixed Price contract against frozen specification
- n First engine run
- n First Engine & Propeller run
- n Flying Test Bed engine
- n First flight of TP400-D6 powered A400M

achieved 28 October 2005

achieved 28 February 2006

- scheduled for early 2007
- scheduled for end November 2007

A military development programme with civil schedule timescales

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TP400-D6 Overview



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TP400-D6 – Technical features

- 1 10,500+ shp
- **3-shaft engine architecture**
- **Overall pressure ratio 25:1**
- **5 Stage IP compressor**
- **1 6 Stage HP compressor**
- **1** Single stage unshrouded HP turbine
- **1** Single stage shrouded IP turbine
- **3 Stage free Power (LP) turbine**
- 1 Chin air intake
- **Offset power gearbox (9.9 ratio)**

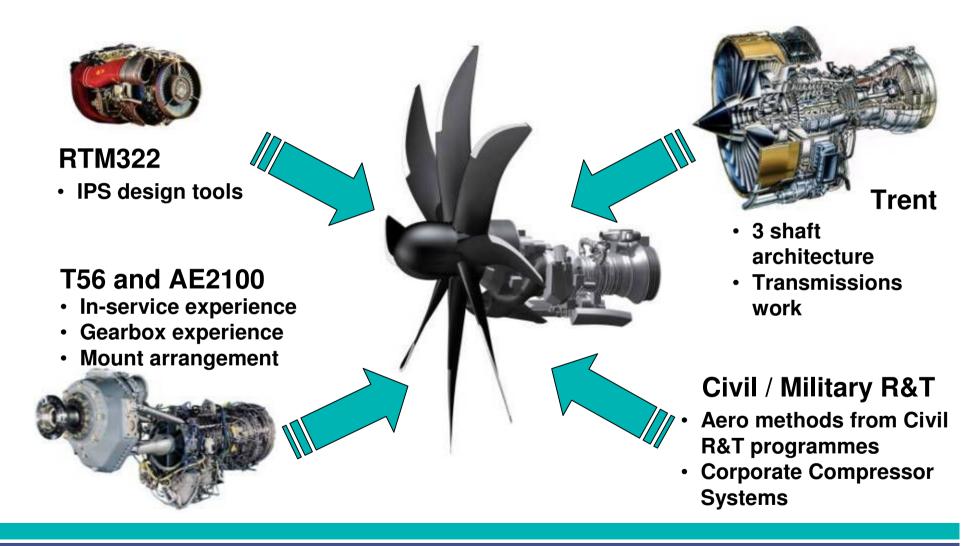


The most powerful modern turboprop in the western world!

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TP400-D6 knowledge transfer examples



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TP400-D6 Validation Programme

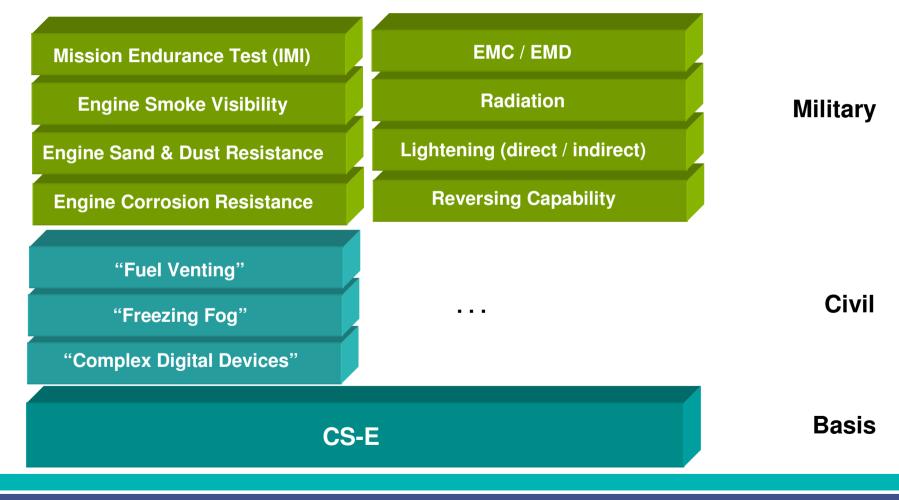


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TP400-D6 Certification basis

- **1** Certification against Civil Basis amended by Military Requirements
- **Qualification** against Airbus Military Specifications



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TP400-D6 Validation programme

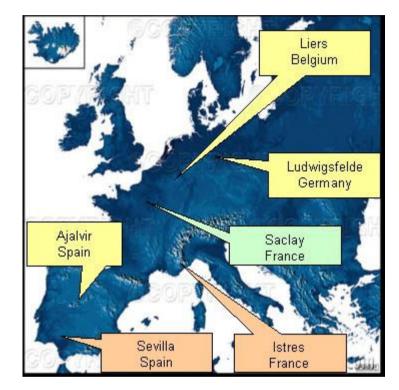
- **3 Sea Level Test Beds**
 - 1 Rain/Hail and Sand/Dust ingestion
 - 1 LP telemetry
 - **150hr type and cyclic endurance test**
 - 1 Thermal paint
 - 1 Overspeed, Overtemperature, etc...

1 Altitude Test Bed

- 1 Performance and handling
- 1 Windmill / relight
- 1 Air/Oil/Fuel System verification
- 1 HP / IP telemetry

2 Outdoor Test Beds

- 1 150hr type tests and cyclic endurance
- 1 Bird ingestion
- **Dyno-prop control and propeller integration**
- **Crosswind and noise**



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TP400-D6 First engine run – 28th October 2005



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First Engine & Propeller Test – 28th February 2005



Snecma Outdoor Testbed (Istres, near Marseille) Engine tested with propeller unfeathered & feathered (up to 15 degrees) Achieved full power

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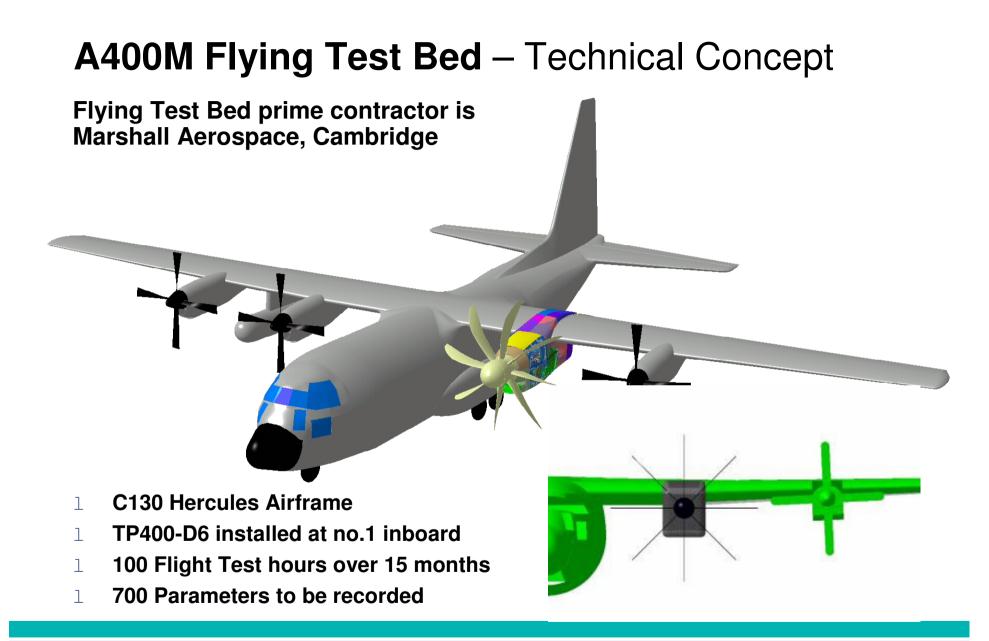
TP400-D6 Flying Test Bed

- 1 The FTB is a C-130 Hercules
 - **1** Owned and operated by Marshall Aerospace of Cambridge.
 - 1 The aircraft is XV108 "Snoopy", the former meteorological research flight aircraft.
- **1** The programme is being led by AMSL
- 1 EPI have to provide an instrumented engine, technical support and a financial contribution.



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Summary



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TP400-D6 – Summary

- **1 TP400 is a major step forward for European Defence** collaborations
 - **Partnership builds on long experience**
 - Commercial practices applied to a military programme
- 1 The TP400 design is a low-risk technical solution tailored to the A400M
 - Optimised 'Rolls-Royce' three-shaft architecture
 - **Designed for 11,000shp and capable of driving 8 bladed 17ft propeller**
 - Component design based on demonstrated technology
 - 1 Robust core with low cycle temperatures
 - Sufficient growth potential

Europrop International combines the best of Europe's engine manufacturers' resources and expertise

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For more news & information http://www.europrop.aero

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