

Hydrogenius

Flying with a Fuel Cell

EWADE 2009, May 14th 2009, Sevilla



Rudolf Voit-Nitschmann
Len Schumann
Steffen Geinitz

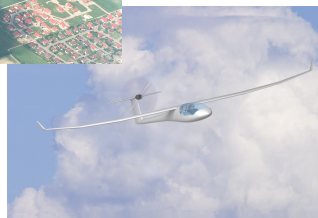
Agenda

- Research in Aircraft Design at IFB
- Berblinger Price 2006 , City of Ulm
- Fuel cell applications in aviation
- Fuel cell technology and components
- Project Hydrogenius
- Outlook

Research: Aircraft Design



Solar aircraft: icaré



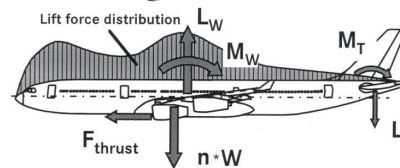
Fuel cell aircraft:
Hydrogenius

Research focal point
„Unmanned Aerial Vehicle“



EU, NACRE
Development
of an IEP

LUFO III HICON,
Weight prediction
for high lift devices



Weight prediction
tools FAME-W,
FAME-FANCY

LUFO III,
„Demonstrator Model“



IFR



Hydrogenius: Team

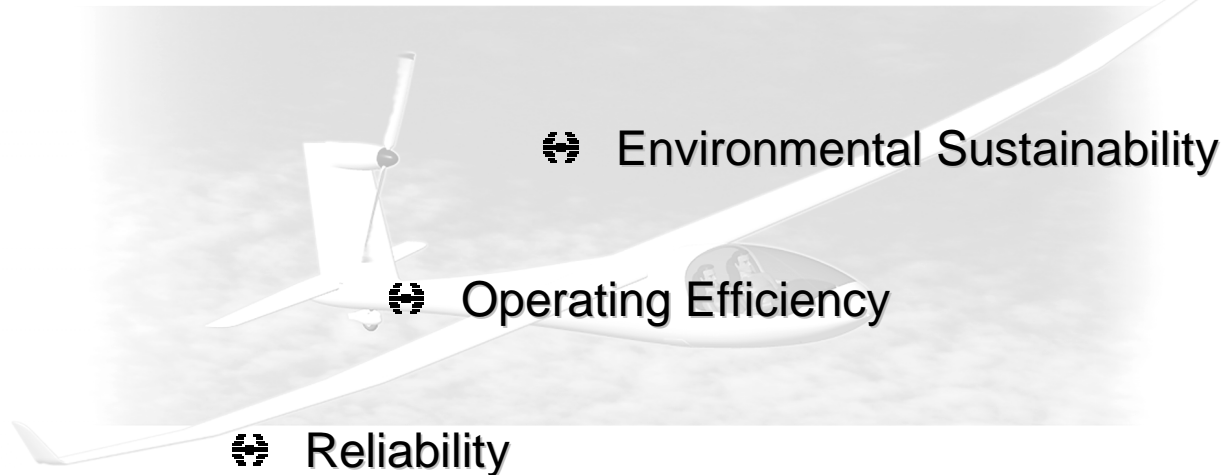
- ⊞ Prof. Dipl.-Ing. Rudolf Voit-Nitschmann:
Project Management and Public Relations
- ⊞ Dipl.-Ing. Len Schumann:
Airframe and Aerodynamics
- ⊞ Dipl.-Ing. Steffen Geinitz:
Propulsion System and Integration
- ⊞ Currently six diploma theses
- ⊞ 11 student assistants for technical documentation,
public relations, simulation and construction
- ⊞ Eight thesis completed related to the project
Hydrogenius



Hydrogenius: The Beginning



Berblinger Preis | 2006



⊕ Light Weight Construction Methods



Hydrogenius: Objectives and Benefit

- ⇌ Flagship initiative for fuel cell technology
 - ⇌ Enormous public effect of aircrafts
 - ⇌ Impressive demonstration of fuel cell performance
- ⇌ Impulses for fuel-efficient mobile applications
- ⇌ Experimental aircraft to investigate fuel cell systems under extreme conditions
- ⇌ Certification of hydrogen and fuel cell technology in aviation
- ⇌ Integration to education at the faculty
 - ⇌ Already six student research projects and diploma thesis are prepared

Hydrogenius: FC Applications in Aviation

Airliner

- ➦ Substitution of Auxiliary Power Unit (APU) and Ram Air Turbine, onboard water generation for facilities etc.
- ➦ Increasing power requirements due to substitution of hydraulic and pneumatic systems by electric systems (more electric aircraft)

UAV & MALE/HALE

- ➦ Low noise operations, low thermal signatures, innovative configurations due to electric propulsion system

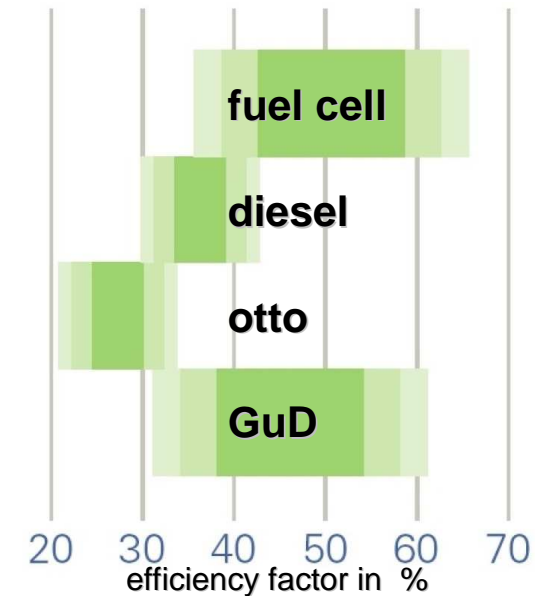
General Aviation

- ➦ Economical, ecological, quiet and comfortable operation, new configurations possible, advanced safety concepts



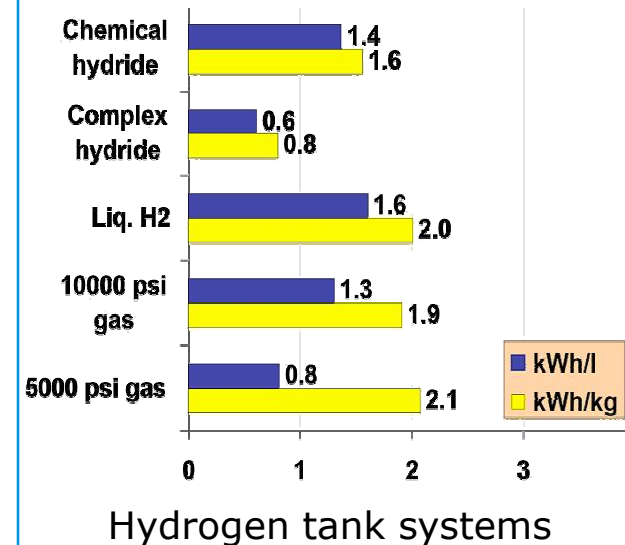
Fuel Cell: Pros and Cons

- ⊕ Zero-emission
- ⊕ High efficiency
- ⊕ Low noise
- ⊕ Vibration-free
- ⊕ Electric drive allows innovative aircraft configurations
- ⊕ Hydrogen can be produced renewable
- ⊖ High amount of weight
- ⊖ High amount of volume
- ⊖ Widespread hydrogen infrastructure is missing yet
- ⊖ Complex hydrogen storage and fueling



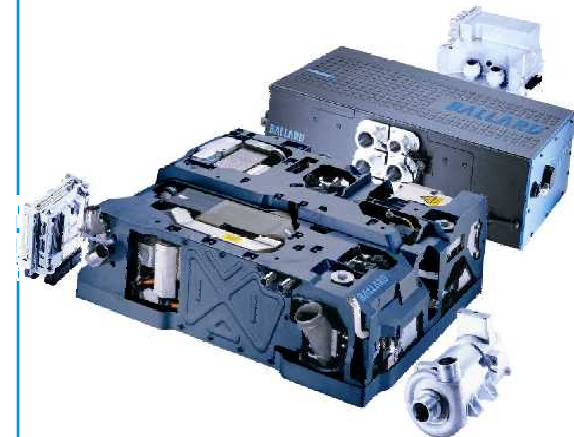
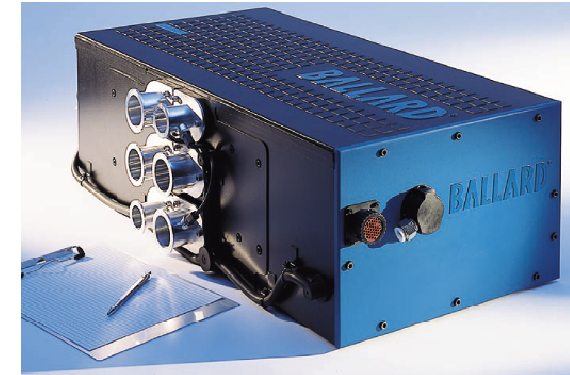
Hydrogen-Storage

- ⊗ Low energy-density of the storage system
- ⊗ High pressure tank is the best solution
- ⊗ Aircraft fuel tank has a capacity of 11 kWh/kg, a hydrogen tank 2,1 kWh/kg
- ⊗ Lithium-polymer-akkus have a spec. Energy ~0,16 kWh/kg up to 0,2 kWh/kg
- ⊗ Without tank we get 12 kWh/kg for conv. fuel and ~ 33,3 kWh/kg for hydrogen
- ⊗ Tank ZM180 by Dynetek (350 bar)
4,2 kg hydrogen at a tank weight of about 93 kg
caloric equivalent to 16,2 l petrol



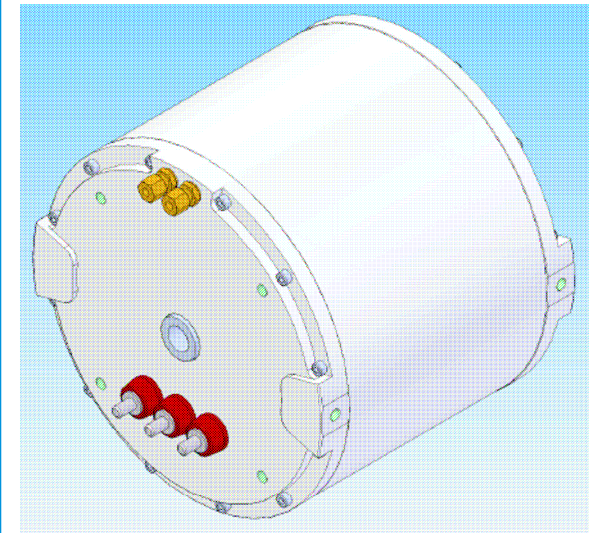
Fuel Cell: System in Preliminary Design

- ⊗ Almost every fuel cell system offering 50-70 kW net power and weighing not more than about 250 kg is applicable
- ⊗ The preliminary design uses as a reference „HY-80™“ fuel cell system of NuCellSys GmbH
 - ⊗ The system delivers 68 kW maximum power and weighs about 220 kg
- ⊗ Currently new fuel cell systems offer more than 85 kW at weights under 200 kg are state of the art



Electric Motor and Power Electronics

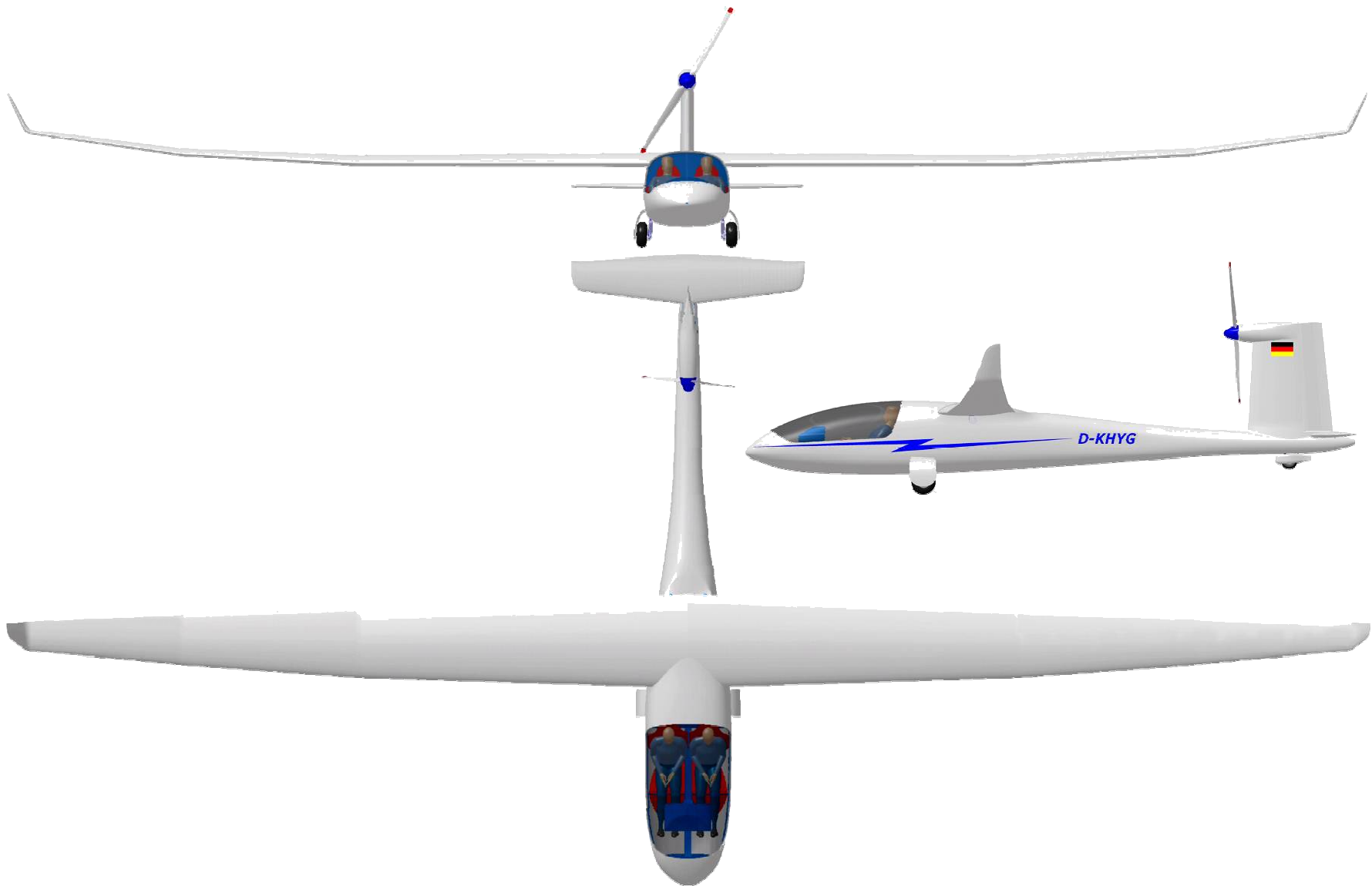
- ⊞ electric motor by **Sineton**
 - ⊞ three-phase brushless permanent magnet synchronous motor
 - ⊞ weight: 25,5 kg
 - ⊞ efficiency: 0,94
 - ⊞ up to 72 kW at 2500 rpm (peak) and 58 kW at 2300 rpm (continuous)
- ⊞ power electronics by **drivetek**
 - ⊞ weight: 8 kg
 - ⊞ Communication via CAN-Bus
 - ⊞ Liquid cooling



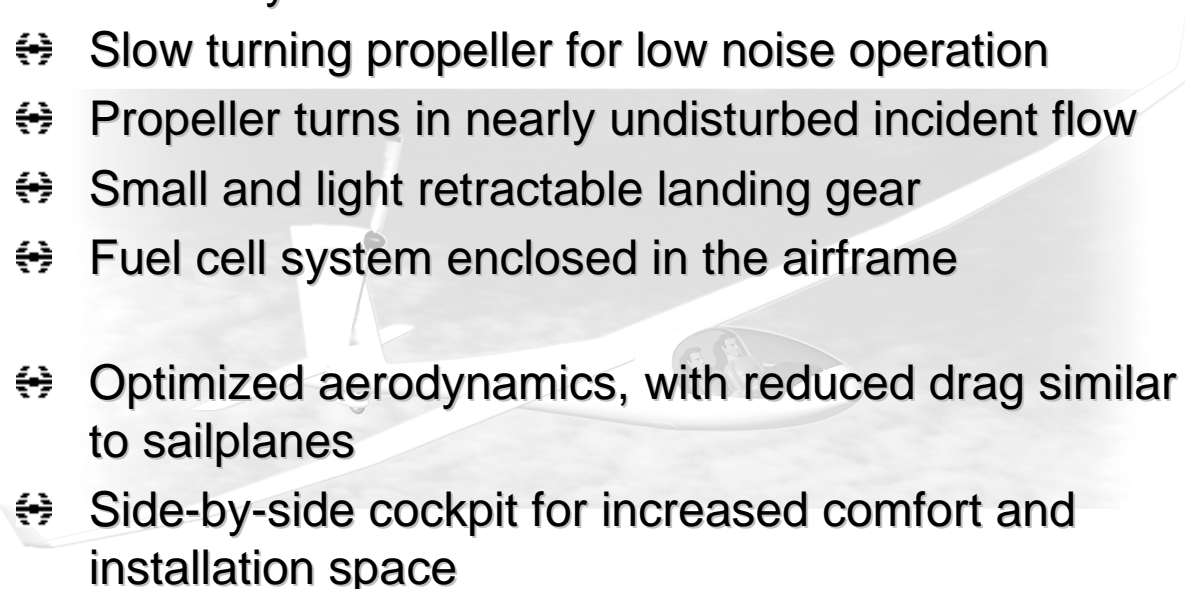
SINETON
RESEARCH AND DEVELOPMENT



drivetek ag
development powerhouse

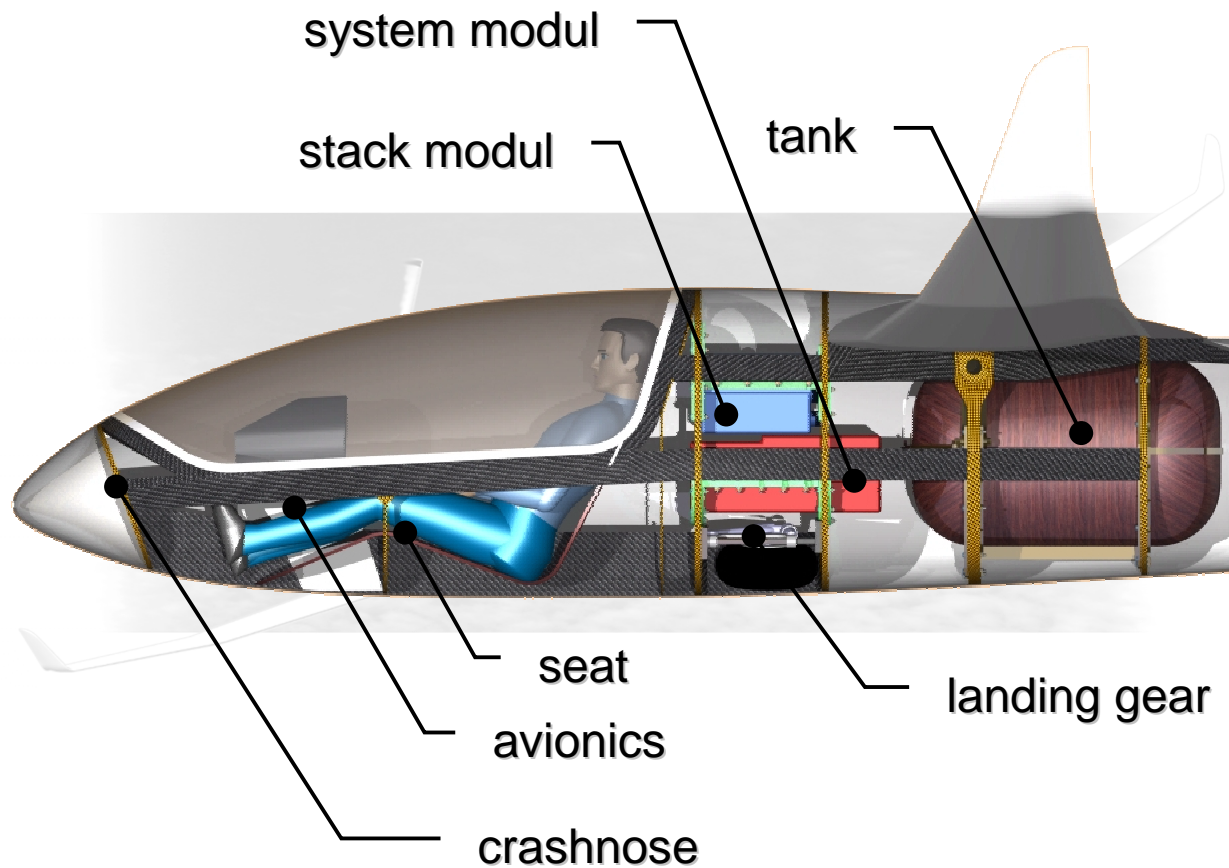


Hydrogenius: Multidisciplinary optimized Configuration

- ⊗ Separation of „power“ and „propulsion“ generation
 - ⊗ Very large variable pitch propeller for increased efficiency
 - ⊗ Slow turning propeller for low noise operation
 - ⊗ Propeller turns in nearly undisturbed incident flow
 - ⊗ Small and light retractable landing gear
 - ⊗ Fuel cell system enclosed in the airframe
- 
- ⊗ Optimized aerodynamics, with reduced drag similar to sailplanes
 - ⊗ Side-by-side cockpit for increased comfort and installation space
 - ⊗ Suitable for daily use

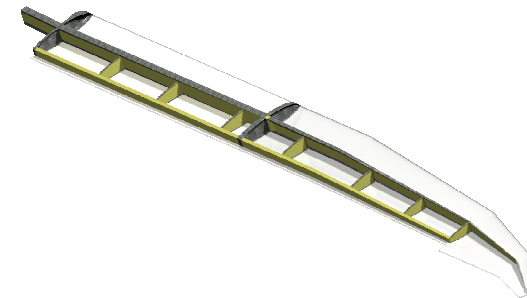


Hydrogenius: System Integration



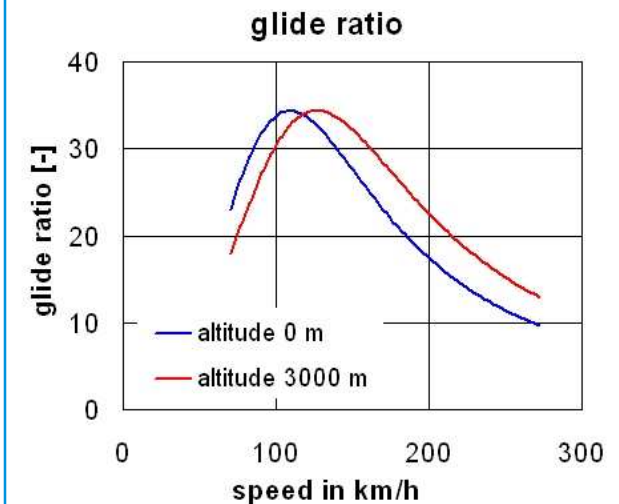
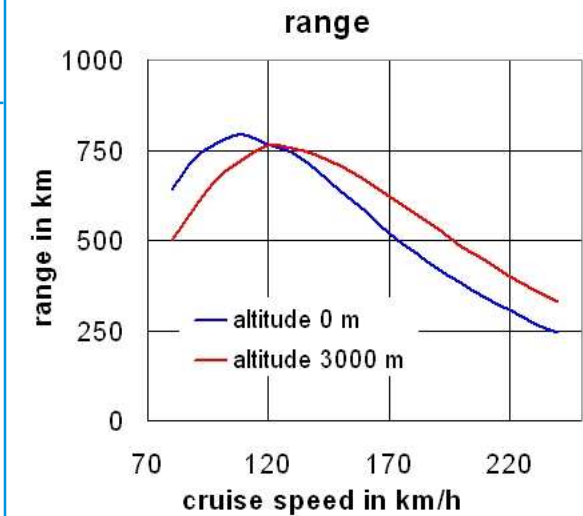
Hydrogenius: Structure and Safety

- ⊕ Side-by-side cockpit
 - ⊕ Large installation space for fuel cell system
 - ⊕ Variably constructed (weight balancing)
- ⊕ Light weight sandwich structures
- ⊕ Safety structure for increased passenger protection
 - ⊕ Use of carbon-/aramid fiber hybrid fabrics for crash energy absorption
- ⊕ Backup battery system
- ⊕ Complete aircraft parachute rescue system

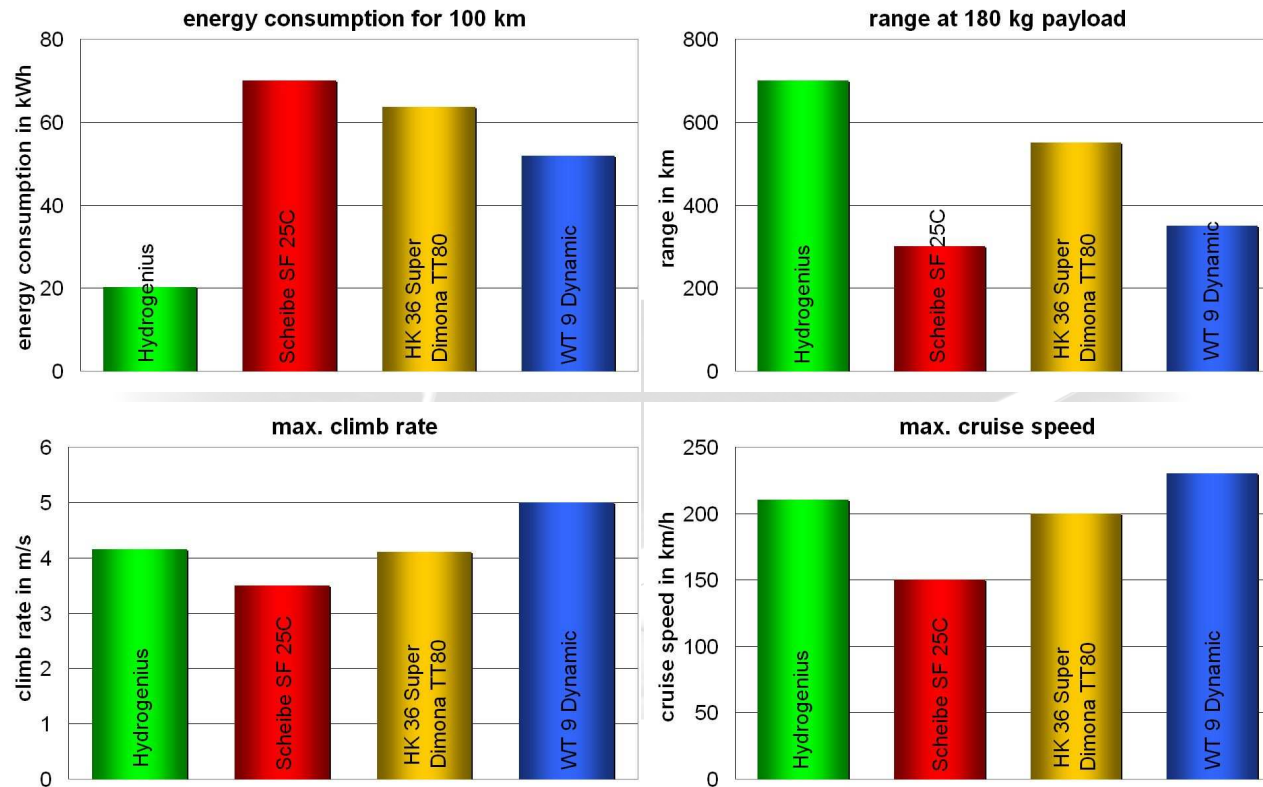


Hydrogenius: Technical Data

⊗ Motor power:	~ 60	kW
⊗ Max. speed:	> 270	km/h
⊗ Best climb rate:	> 4	m/s
⊗ Take-off distance:	< 450	m
⊗ Fuel capacity H ₂ :	4,2	kg
⊗ Service ceiling:	> 4000	m
⊗ Span:	16,86	m
⊗ Wing area:	14,10	m ²
⊗ Wing loading	60,3	kg/m ²
⊗ Length:	8,11	m
⊗ Empty weight:	663	kg
⊗ MTOW:	850	kg
⊗ Payload:	180	kg
⊗ Certification according CS-22		



Hydrogenius: Comparison



Hydrogenius



HK 36 Super Dimona TT80



Scheibe SF 25C



WT 9 Dynamic



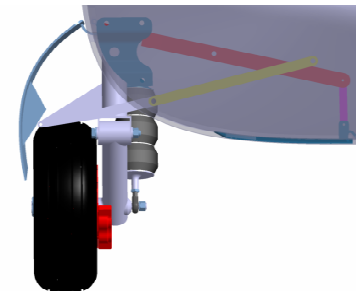
Hydrogenius: Current Status

Technical:

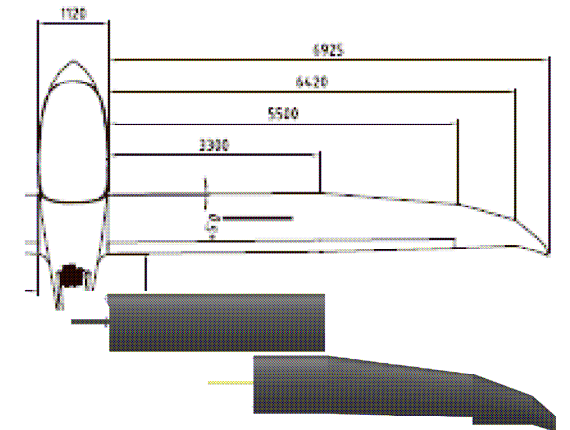
- ➊ Preliminary design is completed
- ➋ Currently: Construction of wing, fuselage, empennage and retractable landing gear in cooperation with PIPISTREL, coolant system, as well as integration of aircraft parachute system

Organizational:

- ➊ Acquisition (necessary funds and tangible means)
- ➋ Acquisition of a fuel cell system
- ➌ Acquisition of the powertrain components
- ➍ Public relations and trade show participation (e.g. AERO Friedrichshafen)



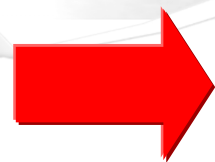
Hydrogenius: Cooperation with Pipistrel



Experiences gained from icaré II

⇌ icaré II has its maiden flight in 1996 and was completely developed and constructed at the faculty of Aerospace Engineering and Geodesy. Thereby knowledge in

- ⇌ unconventional configurations for low energy consumption
- ⇌ light weight structure
- ⇌ electrical propulsion systems
- ⇌ funding via sponsors



icaré II is still the most powerful manned Solar powered aircraft and still in use for research



Hydrogenius: Sponsoring

- ⊗ So far sponsored by nearly 20 companies, institutes, foundations and private persons
- ⊗ Supported by the whole aerospace faculty of the university of Stuttgart with its nine specialized institutes

